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DAVID W TAYLOR NAVAL SHIP RESEARCH AND DEVELOPMENT CE--ETC F/G 13/10
A ROLL, FIN, AND FIN CONTROLLER PREDICTION COMPUTER PROGRAM. (U)
JUN 80 S L BALES, J R TUCKER, G G COX

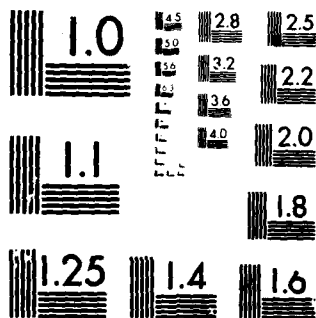
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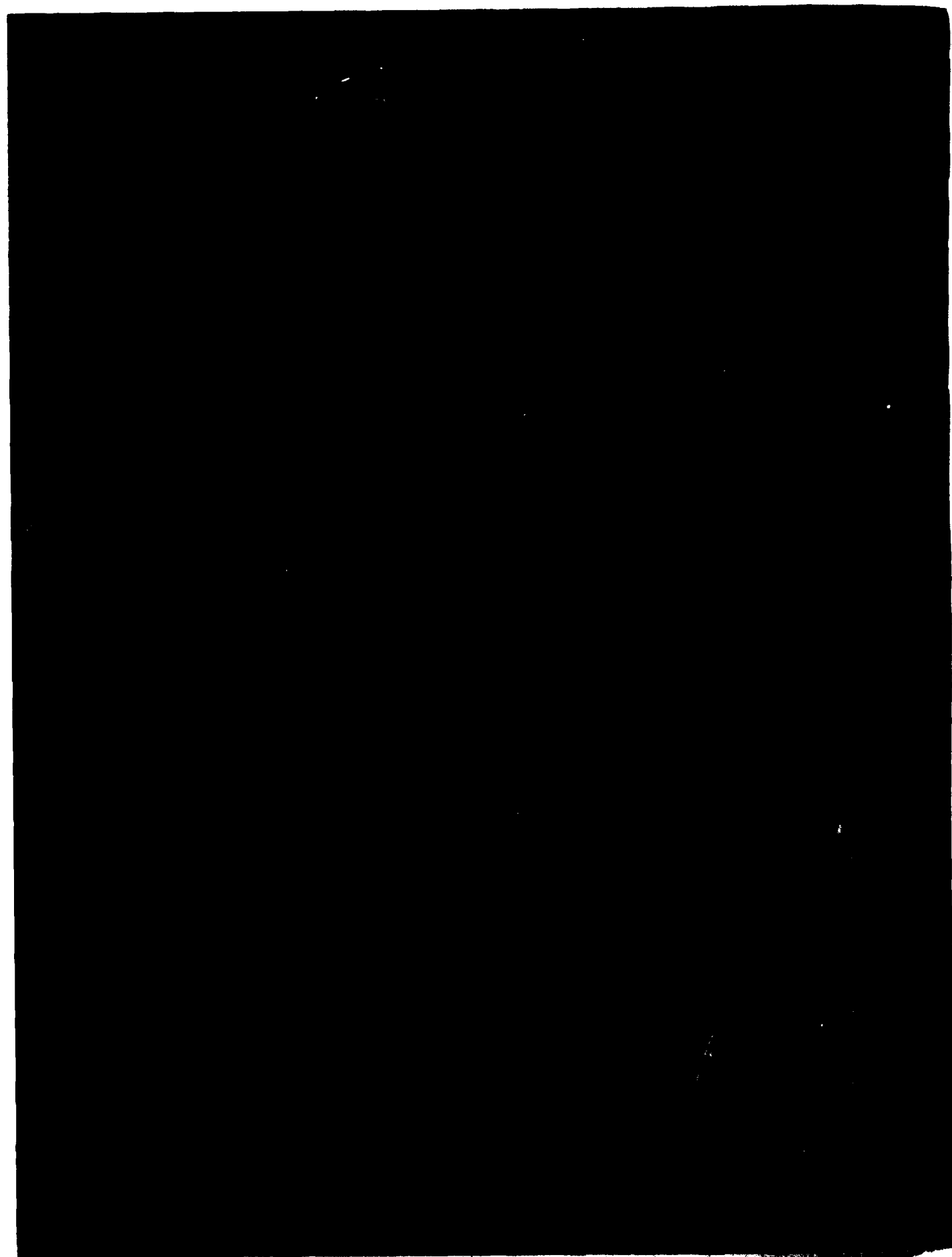
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(Block 20 continued)

bilge keel and antiroll fin sizing effects, and the influence of fin controller characteristics by use of a one degree-of-freedom roll-motion equation. Nonlinear roll damping characteristics, derived from model experiments or by other means, are incorporated by a combination of equivalent linearization and an iteration procedure. Results are predicted for short-crested seas (for stabilizer design purposes) and long-crested seas, which are described by two-parameter Bretschneider wave spectra.

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ABSTRACT

A series of computer programs is under development for use in the design and evaluation of roll stabilization devices. This report is the user manual for the first program that has been completed. Identified by the acronym FINCON, the program is based on the work of Cox and Lloyd published in Volume 85 of the Transactions of the Society of Naval Architects and Marine Engineers. FINCON predicts stabilized and unstabilized ship roll motion, bilge keel and antiroll fin sizing effects, and the influence of fin controller characteristics by use of a one degree-of-freedom roll-motion equation. Non-linear roll damping characteristics, derived from model experiments or by other means, are incorporated by a combination of equivalent linearization and an iteration procedure. Results are predicted for short-crested seas (for stabilizer design purposes) and long-crested seas, which are described by two-parameter Bretschneider wave spectra.

ADMINISTRATIVE INFORMATION

The development and documentation of the computer program reported herein is a part of the Conventional Ship Seakeeping Research and Development Program (Block SF 43 421 202) and the Ship Performance and Hydrodynamics Program (Block SF 43 421 001) both under Program Element 62543. At David W. Taylor Naval Ship Research and Development Center (DTNSRDC) it is identified by Work Units 1504-100, 1507-200, and 1500-104. Authors Susan L. Bales and Geoffrey G. Cox are DTNSRDC personnel. Author John R. Tucker is on the staff of Chi Associates, Inc.

INTRODUCTION

Devices such as bilge keels, anti-roll fins, and anti-roll tanks have been used over the years to reduce the roll motion of naval and commercial vessels. In recent years, the U.S. Navy has become increasingly involved in the design and development of suitable roll stabilization devices for Navy ships. This report provides a user's manual for a computer program which permits prediction of unstabilized and bilge keel/anti-roll, fin-stabilized, ship roll motions; bilge-keel and fin-sizing requirements; and the influence of fin controller characteristics. The program is known by the acronym FINCON.

The need for improved design and performance evaluation tools for bilge keels, antiroll fins, and their controllers has been recognized by Cox and Lloyd,^{1*} who provide the hydrodynamic basis for such investigations. A comprehensive work, Reference 1 covers such topics as the current state-of-the-art for bilge keels, fins and tanks, the status of lateral motion predictions, measures of effectiveness, design practices, and sea state specifications for design. The computer program described in this report is based on the procedures detailed in Reference 1.

Reference 1 recognizes the potential need for roll and roll stabilization prediction tools throughout the so-called design spiral of U.S. Navy ships. Hence, the procedures outlined there and used here can be employed with a very simple descriptor of ship particulars and geometry. Specifically, modifications to the one-degree-of-freedom roll motion equation of Conolly² are used to predict ship roll motion, and the required program input is rather easy to obtain from the data available during early stages of ship design. For instance, in addition to specification of sea condition and ship speed, particulars such as ship length, beam displacement, transverse metacentric height and radius, natural roll period, and roll decay coefficients are required. The estimates of these required input variables can be refined as the design process continues and more accurate data becomes available through model experiments, etc. The manpower and computer time costs involved in such predictions are relatively low, and these predictions can be completed quickly in comparison to other sea-keeping design evaluation procedures.

An added feature of the FINCON program is the capability to recognize the effect of fin saturation, which occurs in heavy seas, on RMS (root mean square) roll angle. The approach is based on a refinement of the method given in Appendix 3 of Reference 1, and specific details of this improvement will be published in a future report currently under preparation by Cox.

*A complete listing of references is given on page 59.

PROGRAM ORGANIZATION

FINCON, written in extended FORTRAN, is operable on DTNSRDC's CDC 6000 computers.

Figure 1 is an overview of the organization of FINCON. The overall program consists of two major parts, FINCON and FINSTAB with FINCON acting as the driver or so-called main program.

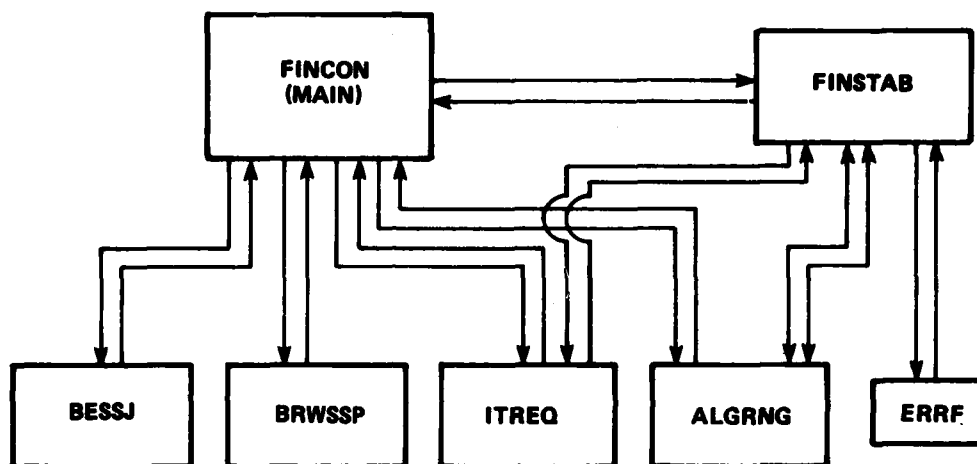


Figure 1 - Program Organization

As the driver of the rest of the program, FINCON calls all other routines for execution. In addition, FINCON provides the unstabilized roll calculations (including the effect of bilge keels) and is responsible for all program input and output.

FINSTAB is the second major part of the program. It predicts fin-stabilized roll angles, fin angles, fin angular velocities, and fin angular accelerations for the input fin and fin controller characteristics. FINSTAB also includes the fin saturation calculations. Although it is called into execution by FINCON, given the proper input and output specifications, FINSTAB in reality can be executed as an independent program.

BESSJ is a CDC system subroutine which is called up at the time of execution for calculation of Bessel functions of the first kind. BESSJ is only used when the ship's waterplane is declared elliptical.

BRWSSP is a subroutine to predict the Bretschneider wave-slope spectrum for a specified significant wave height and modal wave period. Initially, the wave-height spectrum is computed; and then it is converted to a wave-slope spectrum by multiplying by the product of a constant and the square of the wave number k ,

$$k = \omega^2/g \quad (1)$$

where ω is the wave frequency in radians per second and g is the acceleration due to gravity. The constant $(180/\pi)^2$ enters in to permit conversion into degrees to yield values of the roll-response amplitude operator, which is computed by FINCON in units of $(\text{degree/degree})^2$.





ITREQ is a subroutine which iterates between an equivalently linearized roll damping curve, which is a function of RMS roll rate, and the predicted RMS roll rate. In brief, the iteration continues until the computed short- or long-crested RMS roll rate, either unstabilized or stabilized, is within a small value, epsilon, of the previously computed value. The appropriate RMS roll angle can then be found. Additional details of the exact procedure are provided in the Appendixes A, B, and C. ITREQ is called by both FINCON and FINSTAB.

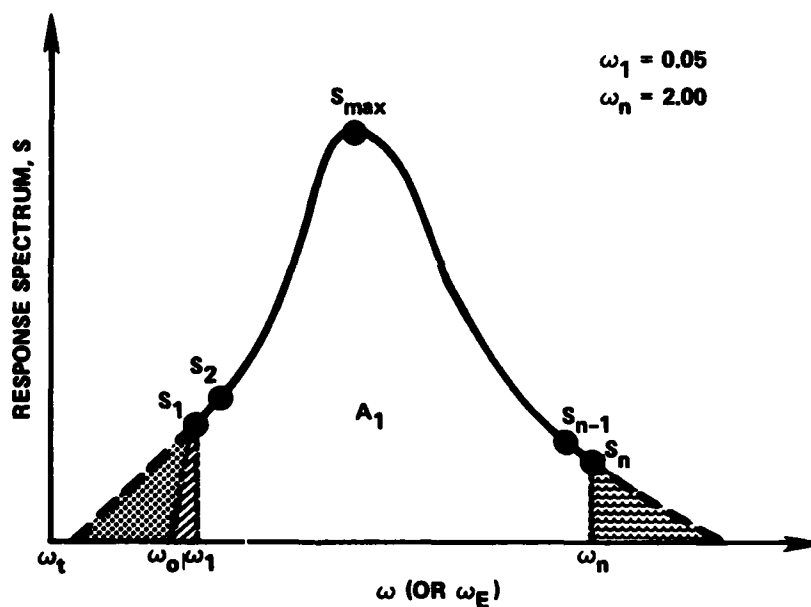
ERRF is a function subprogram which gives a rational approximation to the error integral

$$\text{erf}(\chi) = \frac{2}{\sqrt{\pi}} \int_0^\chi e^{-t^2} dt \quad (2)$$

and is used in the calculation of fin saturation effects.

ALGRNG is an integration subroutine which performs a so-called Lagrangian or quadratic integration over three points at a time. The subroutine is called by both FINCON and FINSTAB, and spectral closure is ensured by the techniques outlined in Figure 2, adopted from Reference 3.

-  A_1 = AREA OF COMPUTED RESPONSE SPECTRUM
-  $A_2 = 1/2 S_1 (\omega_1 - \omega_0)$ WHERE $\omega_0 = \omega_1 - 0.03$
-  A_3 = AREA OF RIGHT TRIANGLE FORMED BY DRAWING STRAIGHT LINE THROUGH S_1 AND S_2 TO THE ABSCISSA, TRIANGLE $\omega_t S_1 \omega_1$
-  A_4 = SAME AS A_3 BUT FOR S_{n-1} AND S_n



IF $S_1 > 0.1 S_{max}$, THEN $A_1 \leftarrow A_1 + \text{MINIMUM OF } A_2 \text{ AND } A_3$
 IF $S_n < S_{n-1}$ AND $S_n > 0.1 S_{max}$, THEN $A_1 \leftarrow A_1 + A_4$

Figure 2 - Spectral Closure Procedure
 (From Reference 3)

The procedure for predicting the unstabilized and stabilized roll angles, as well as fin angles, and velocities is outlined in Appendix A. Because the procedure is developed in detail in Reference 1, only a listing of the equations used is given in Appendix A. In general, the FORTRAN variables have been named as closely as possible to the variables of the equations of Appendix A, so the user should have relatively little difficulty in "reading" the program. A listing of the program is given in Appendix B. Appendix C describes two special algorithms used, namely the iteration algorithm and the cosine squared short-crested sea algorithm for arbitrary spreading angles.

PROGRAM INPUT

As described in the Introduction, the input requirements of FINCON are relatively simple. The input variables consist of sea conditions, ship speed, and very simple descriptions of the ship particulars and geometry. Unlike many multi-degree-of-freedom ship-motion prediction programs in use today, neither offsets of the ship sections nor Lewis forms are required. This makes implementation of FINCON possible at an early design stage (e.g., before the ship lines are "firmed up").

Table 1 describes the required input to the program and Table 2 explains the notation and variable names. Up to eight sea conditions and five ship speeds may be executed within a single run, using either metric or English units for input/output. If ITERATE on card 10 is greater than zero, then roll damping is treated as nonlinear, and coefficients, DUC, should be input. For example, letting $d = \text{DUC}$ for a simpler notation, then

$$n = d_0 + 1.61d_p y^p + 1.88d_1 y + 4d_2 y^2 + 9.4d_3 y^3 + 24d_4 y^4 \quad (3)$$

where $y = \sigma_\phi / \omega_\phi$ is the RMS roll rate divided by ship natural frequency, p is 0.772, and n is the roll damping coefficient as a function of y .

TABLE 1 - INPUT TO FINCON PROGRAM

Card	Contents	Format
1	NAME1, NAME2, NAME3	(3A10)
2	TITLE2	(8A10)
3	NWH, NV, IUNITS	(3I5)
4	(SWH(IWH), IWH=1, NWH) where $1 \leq NWH \leq 10$	(8F10.5)
5	(TO(IWH), IWH=1, NWH) where $1 \leq NWH \leq 10$	(8F10.5)
6	(VK(IV), IV=1, NV) where $1 \leq NV \leq 5$	(5F10.5)
7	DISPTON, L, T, GM, BM, TPHI, Q	(7F10.5)
8	SHAPE	(A10)
9	ISC, ANGLE	(15,F10.5)
10	ITERATE	(I5)
11	IF ITERATE = 0 ((DUC(IV,1), IV=1, NV) where $1 \leq NV \leq 5$	(5F10.5)
or		
11	IF ITERATE \neq 0 ((DUC(IV,1), I=1, 6), IV=1, NV) where $1 \leq NV \leq 5$	(6F10.5)
12	IPRINT(1), IPRINT(2)	(2A10)
13	NSTAB	(I5)
Repeat cards 14 through 17 NSTAB times.		
14	M, AREA, R	(15,2F10.5)
15	DCLDBFS, H0, H1, H2, H3, H4	(6F10.5)
16	GK, GV, K1, K2, K3	(5F10.5)
17	A1, A2, A3, B1, B2, B3	(6F10.5)
18	NSAT	(I5)
19	(BSTOP (I), BVELMAX (I), I=1, NV)	(8F10.5)

TABLE 2 - PROGRAM NOTATION (INPUT)

ANGLE	Increment of wave energy spreading for calculation of short-crested responses; e.g., 5, 10, or 15 degrees
AREA	Fin area, square feet or square meters
A1, A2, A3	Fin servo coefficients
BM	Transverse metacentric radius, (distance between center of buoyancy and metacenter), feet or meters.
B1, B2, B3	Fin controller compensation coefficients
BSTOP	Fin limit angle
BVELMAX	Fin limit angular velocity
DCLDBFS	Free stream lift coefficient curve slope, per degree
DISPTON	Ship displacement, long tons or metric tons
DUC	Roll damping or roll damping coefficients
GK	Overall fin control gain
GM	Transverse metacentric height, feet or meters
GV	Speed dependent fin control gain
H0, H1, H2, H3, H4	Fin lift curve correction coefficients
IPRINT	Array of print options; IPRINT(1) = SPECTRA, heading printing of long-crested response spectra and components. IPRINT(2) = ITERATN, step by step printing of iteration over roll damping of relevant variables.
ISC	Switch for short-crested responses. If ISC \neq 0, provide ANGLE.
ITERATE	Equals 0 for roll damping value independent of roll angle. Equals 1 when iteration is required.
IUNITS	Switch indicating type of units for Input/Output: IUNITS = 0 for English, = 1 for metric.
K1, K2, K3	Roll angle, velocity, and acceleration characteristic gain factors. Sensitivities of demanded fin angle to roll angle, velocity, and acceleration.
L	Ship length between particulars, feet or meters

TABLE 2 (Continued)

M	Number of fin pairs
NAME1, NAME2, NAME3	Identification of run of program by person's name, code, and telephone extension
NSAT	Equals 1 for inclusion of significant saturation effects
NSTAB	Number of fin/controller/servo, etc., input sets; e.g., ≤ 10
NV	Number of ship speeds, e.g., ≤ 5
NWH	Number of sea conditions, e.g., ≤ 10
Q	$\delta I / (I + \delta I)$, see Equation (5) for definition of $\delta I / I$
R	Fin moment arm, feet or meters
SHAPE	Alphameric description of waterplane shape; e.g., PARAB, ELLIP, or RECTANG
SWH	Significant wave height, feet or meters
T	Draft, feet or meters
TITLE2	Alphameric identification of a run of the program; e.g., ship name
TPHI	Roll period, seconds
TO	Modal wave period, seconds
VK	Ship speed, knots

The d_j values are found by fitting a curve* to experimentally derived calm water roll decay data, or by use of analytically predicted values, prior to execution of FINCON. The iteration is performed until

$$|1 - (\hat{\sigma}_{\phi}^2 / y)| \leq 0.01 \quad (4)$$

where $\hat{\sigma}_{\phi}$ is that RMS roll rate obtained from the prediction using the roll damping coefficient n associated with y by Equation (3).

*In practice, a straight line frequently provides an adequate representation of the experimentally derived roll data.

Card 12 defines two print options. If IPRINT(1) is SPECTRA, a heading-by-heading printing of long-crested roll spectra and their components (e.g., the wave frequencies ω , the response amplitude operators (RAO's), etc.) will be printed. If IPRINT(2) is ITERATN, step-by-step printing of the iteration over roll damping of the relevant variables will occur. This option is useful for debugging purposes.

If NSTAB is zero on card 13, only unstabilized roll predictions will be made and cards 14 to 17 can be eliminated. If NSTAB is greater than zero, cards 14 to 17 should be repeated NSTAB times.

Card 18 specifies whether or not saturation calculations will be done. If NSAT is zero (blank card) no other cards are needed. If NSAT is greater than zero, NSAT pairs of values for fin limit angle (BSTOP) and fin limit velocity (BVELMAX) are needed. If BVELMAX is not supplied (field left blank) a value will be generated by FINCON (e.g., $BVELMAX=10 \cdot BSTOP/TPHI$).

Table 3 gives a sample listing of input cards for an example ship, between the END OF RECORD and END OF FILE cards. It should be noted that the input values are given in units of feet and long tons. A metric conversion option may be invoked to allow the input of values in metric units. The method for activating this option is to place a "1" in Column 15 (IUNITS) of card 3.

The first two data cards shown in Table 3 contain only alphanumeric, or descriptive data. Card 3, indicates that two sea conditions and one speed are to be considered and that English units are assumed for Input/Output. Cards 4 and 5 define the sea conditions in terms of significant wave height and modal wave period. Card 6 specifies the ship speed at 25 knots. Card 7 gives the ship particulars of displacement, length, draft, transverse metacentric height, transverse metacentric radius, roll period, and Q. Q is the ratio of added mass to total mass moment of inertia, $\delta I/(I+\delta I)$, and as shown in Reference 1, can be estimated from

$$\frac{\delta I}{I} = -0.186 + 1.179 C_B - 0.615 C_B^2 \quad (\text{without bilge keels})$$

and

(5)

$$\frac{\delta I}{I} = -0.002 + 0.814 C_B - 0.316 C_B^2 \quad (\text{with bilge keels})$$

TABLE 3 - TYPICAL CONTROL AND DATA CARD SET

```

CHHJZXX,CM77000,T199,P4.
CHARGE,CHHJ,HQHAA15023,RS.
ATTACH,OLDPL,FINCONREVISION6,ID=PUAA.
MAP(ON)
SETCORE(INDEF,ADDR)
ATTACH(NSRDC)
LIBRARY(NSRDC)
FTN,I=OLDPL.
LGO.
7/8/9  END OF RECORD
Card 1: H.JONES      1568      71210
2: CGN-42  SELECTED FINS (TWO FIN PAIRS, 75 SQ.FT.) R. N. O. (25 KTS)
3:      2      1      0
4:      24.61      18.04
5:      12.9      12.3
6:      25.0
7: 12000.0      560.0      22.7      4.3      16.06      12.8      0.331
8: PARAB
9:      1      15.0
10:      1
11: 0.1693      0.00570      -.0002
12:
13:      1
14:      2      75.0      33.45
15: 0.43      0.349      1.117      -0.519
16: 1.0      1.650      1.0      2.5      1.0
17: 1.0      0.160      0.025      1.0      0.630      0.092
18:      1
19: 10.2      7.98875
6/7/8/9  END OF FILE

```

where C_B is the block coefficient and average bilge keels are assumed. Card 8 specifies waterplane shape as parabolic, which is usual for fine-form naval ships. Card 9 indicates that short-crested seas will be treated and specifies the spreading angle to be 15 degrees. If card 9 were blank, the program would assume that only long-crested calculations would be done. Card 10 indicates that iteration over the roll-roll damping curve is required to account for nonlinear roll damping. Card 11 gives the coefficients, DUC or d, of the roll-roll damping curve (e.g., see Equation (3)). Had roll damping been linear in this sample input case, card 10 would have

been blank and card 11 would have contained the single value for roll damping coefficient, n , for 25 knots. Card 12 is blank, so printing of any intermediate steps (e.g., the long-crested spectra and their RAO's, etc., or the iteration steps) is not done.

Cards 1 to 12 provide all the data necessary to complete a FINCON run to calculate unstabilized roll angles. If card 13 were blank, this is exactly the way the program would execute. However, card 13 indicates that one set of stabilizing conditions is to be considered. Cards 14 to 17 provide the fin and fin controller particulars. Two pairs of 75 square feet fins with a fin moment arm of 33.45 feet are indicated on card 14. The fin moment arm is taken about the longitudinal axis, through the center of gravity, and measured to the center of the fin for all fin pairs. Card 15 gives the free stream, lift coefficient curve slope and the lift correction coefficients which compensate for fin-induced sway and yaw motions. The values of card 15 are estimated by the procedures outlined in Reference 1. Card 16 specifies both the overall gain G_K and speed dependent fin control gain G_V , as well as the roll angle, roll rate, and roll acceleration characteristic gains k_1 , k_2 , and k_3 , such that the fin controller equation is

$$\beta = G_K \cdot G_V (k_1 \phi + k_2 \dot{\phi} + k_3 \ddot{\phi}) \quad (6)$$

where ϕ is the roll angle and β is fin angle. Card 17 specifies the fin servo coefficients and the fin-controller compensation coefficients. The values given in Table 3 for card 17 variables were selected from Reference 4 and are also given in Reference 1.

Card 18 is not blank, indicating that any significant saturation effects will be included in the calculations, and so an additional card is required to provide the limiting angle and limiting speed* of the fins. Insignificant saturation effects are automatically ignored by the program.

*If the angle is provided and the speed is left blank, the program will compute an appropriate value.

This is to avoid the additional cost which would otherwise be incurred while having no significant effect on the results. Had card 18 been blank, indicating that no saturation effects were to be considered, then no other data cards would have been needed.

PROGRAM OUTPUT

Table 4 presents the program output for the sample input given in Table 3. The first page outputs the input identifying titles; the second page outputs the operating conditions and ship/fin/fin controller particulars; and the third page provides the results. The first listings on the third page gives the resulting unstabilized RMS roll angles and corresponding damping coefficient values for ship headings from 0 to 180 degrees (following to head seas) in 15-degree increments for short-crested seas. The next row of tables contains the corresponding RMS stabilized roll angles and damping coefficients, as well as the resulting RMS fin angles and velocities. Those values of RMS roll for which saturation effects would produce less than a 2 percent change are indicated by an asterisk. For such headings, only the unsaturated values are calculated.

Had there been a second stabilized condition specified on input card 13, another row of tables would follow the one for case 1. Results for additional speeds and sea conditions would be printed in a similar fashion on subsequent pages. The fourth page indicates that the program completed execution satisfactorily; e.g., no system (loader, input/output, etc.) errors were encountered.

Table 5 presents a typical output when IPRINT(1) is SPECTRA on card 12. One such page would appear for each ship heading. The columns provide wave frequency W, wave-encounter frequency WE, wavelength LAM, wavelength-to-shiplength LAM/L, wave number K, nondimensional transfer function TR, nondimensional response amplitude operator, RAO, wave-slope spectrum W SL S, and roll (unstabilized) response spectrum SUR. Also given are the dimensional response amplitude operator RAO DIM, the wave-height spectrum W HT S, and the resulting roll response spectrum SUR DIM, which should be equivalent to SUR. Due to the fact that this sample is for the case of nonlinear

TABLE 4 - TYPICAL PROGRAM OUTPUT, ITERATION OVER ROLL DAMPING

* * * ROLL MOTION PREDICTION PROGRAM * * *

W.JONES 1568 71210

TABLE 4 (Continued)

CGM-02

SIGNIFICANT WAVE HEIGHT(S) (FEET) = 24.01 18.06
 WAVE PERIOD(S) (SECONDS) = 12.90 12.30
 SHIP SPEED(K) (KNOTS) = 25.0
 DISPLACEMENT (L. TONS) = 10000.
 LENGTH BETWEEN PP (FEET) = 500.0
 DRAFT (FEET) = 22.70
 TRANSVERSE METACENTRIC HEIGHT (FEET) = 4.3
 METACENTER ABOVE BUOYANCY CENTER (FT) = 10.06
 ROLL PERIOD (SECONDS) = 12.00
 0 = .431
 WATERPLANE SHAPE = PARAB
 SPREADING ANGLE = 15.

ITERATION OVER ROLL DAMPING WILL BE DONE.

DAMPING INPUT IN THE FORM $n = C1 + 1.01eC2y + 1.00eC3y + 4.00eC4y + 9.60eC5y + 24.00eC6y$

SPEED (KNOTS) C1 C2 C3 C4 C5 C6
 25.0 .1053 0.0000 6057 -.0002 .0000 0.0000

ROLL STABILIZATION WILL BE CALCULATED FOR 1 CASES

FIN AND CONTROL SYSTEM PARAMETERS ARE AS FOLLOWS:

CASE	N	A	H	(UCL/UBIFS)	MU	M1	M2	M3	M4	M5	K1	K2	K3	A1	A2	A3	M1	M2	M3	
		FT	50	FT	PER DEG															
1	2	75.00	33.45		.063	.349	1.117	-.519	0.000	0.000	1.000	1.000	2.500	1.000	.100	.025	1.000	.630	.092	

IV RSTUP OVELMAX
 1 10.20000 7.90075

TABLE 4 (Continued)

CON-42

SIGNIFICANT WAVE HEIGHT = 44.61 FEET
 MODAL WAVE PERIOD = 12.40 SECONDS
 SHIP SPEED = 25.0 KNOTS

UNSTABILIZED RMS ROLL (DEGREES)

HEADING	N	SC
0	.1830	2.30
15	.1850	2.48
30	.1892	2.97
45	.1923	3.23
60	.1943	3.44
75	.1952	3.40
90	.1949	3.20
105	.1935	2.92
120	.1910	2.41
135	.1874	1.82
150	.1832	1.26
165	.1797	.85
180	.1763	.70

CASE 1: STABILIZED RMS ROLL (DEGREES)

HEADING	N	SC
0	.1770	1.36
15	.1789	1.44
30	.1815	1.62
45	.1840	1.79
60	.1858	1.90
75	.1868	1.91
90	.1870	1.91
105	.1863	1.82
120	.1848	1.36
135	.1826	1.06
150	.1800	.78
165	.1778	.57
180	.1769	.50

FIN MOTION (DEGREES)

SC
2.64
2.88
3.40
3.94
4.32
4.49
4.42
4.10
3.58
2.92
2.24
1.71
1.51

FIN VELOCITY (DEGREES/SECOND)

SC
.88
1.11
1.57
2.05
2.45
2.74
2.88
2.84
2.65
2.32
1.92
1.57
1.43

• SATURATION FACTOR IS INSIGNIFICANT

TABLE 4 (Continued)

*** END ***

roll damping, the spectral data and the RMS roll of Table 5 are not, in general, considered especially meaningful. Had the damping been linear, as is usually the case for the ship without bilge keels, the spectra would be representative of the long-crested seas case and the RMS roll could correspond to a value given on the "page-three-type-output" of Table 4.

Table 6 presents a typical output when IPRINT(2) is ITERATN on card 12. In brief, the intermediate roll rate and damping coefficient values are printed for each heading, speed, and sea condition. Appendix C further describes this output example.

TABLE 6 - TYPICAL PROGRAM OUTPUT FOR IPRINT(2) = ITERATN OPTION

IV	IMU	NTRY	PHIN	YP+1	YP	YP-1	GP	GP-1
1	7	1	.094	.475	0.000	.239	4.628	.476
1	7	1	.094	.475	2.314	0.000	4.628	4.628
1	7	2	.140	.475	2.314	0.000	3.847	4.628
1	7	2	.140	3.461	3.461	2.314	3.847	3.847
1	7	3	.156	3.461	3.461	2.314	3.674	3.847
1	7	3	.156	3.645	3.645	3.461	3.674	3.674
1	7	4	.159	3.645	3.645	3.461	3.652	3.674
1	7	1	.094	3.645	0.000	3.461	.507	3.674
1	7	1	.094	3.645	.254	0.000	.507	.507
1	7	2	.100	3.645	.254	0.000	.506	.507
1	7	2	.100	.504	.504	.254	.506	.506
1	7	3	.105	.504	.504	.254	.504	.506

PROGRAM EXECUTION

A typical deck control card set up is given in Table 3. Simply speaking, the object program is attached and executed. The object program is stored permanently on a private disk pack and can be recovered for storage on the main (public) disk and for user execution by running the control card deck of Table 7. The source deck is also stored on the private disk pack in an UPDATE file such that program modifications can be easily made, if necessary. The program listing of Appendix B was printed from this UPDATE file.

TABLE 7 - CONTROL CARD SET TO RETRIEVE OBJECT PROGRAM

COLS. 123456789112345678921234567893123456789412345678951234567896123456789

CHHJPAK,CM77777,T100,RP1,P3.
 CHARGE,CHHJ,XXXXXXXXXX,CC,R.
 PAUSE. JOB REQUIRES DISK PACK DV4850.
 MOUNT,VSN=DV4850,SN=HJPKL4.
 REQUEST,TWO,*PF.
 ATTACH,ONE,FINCONOBJECTNOV,ID=CHHJ,CY=1,MR=1,SN=HJPKL4.
 COPYBF,ONE,TWO,1.
 CATALOG,TWO,FINCONOBJECTNOV,ID=PUAA,AC=XXXXXXXXXX,CY=1,MR=1.
 6/7/8/9 END OF FILE

The run time of the program, indicated by TXXX on the job card of Table 3, varies, of course, with the amount of calculation required. Roughly, for nonlinear roll damping and 15-degree short-crested spreading, a stabilized roll calculation (without saturation effects, and for a single sea condition and speed) takes about 50 seconds of execution time and 15 seconds compilation time. For unstabilized calculations, the execution time is somewhat less than half of the time for the stabilized case. The time increases proportionately as the spreading angle of the short-crested seas is decreased. For a 5-degree spreading angle, the time is almost three times (~ 145 seconds) that of the 15-degree case (reflecting the fact that there are about three times as many calculations that need to be performed). From the runs made to date, it is not evident that decreasing the spreading angle from 15 degrees increases the accuracy of predicted roll at a given ship heading by a noticeable amount. However, a finer mesh of spreading angles does, in some cases, permit a more refined localization of the worst heading angle. Thus, the required execution time is a multiple of 50 seconds depending on the number of speeds, sea conditions, and the value of the spreading angle (in proportion to 15 degrees).

The required memory, as indicated by CMXXXXXX on the job is 77777 octal words (see Table 3). The job priority, indicated by PX on the job card, is then determined by the amount of system time required. Based on current computer center figures for the CDC 6700 and average costs over several program runs, the guidelines in Table 8 are offered.

TABLE 8 - RUN TIME AND COST GUIDELINES

T	Highest Priority (Turnaround)	~ \$/System Seconds
< 200	4 (prime shift, 1 hour max after completion)	0.090
<3600	3 (prime shift, as soon as possible)	0.074
Unlimited	2 (nonprime shift, when possible, overnight)	0.060

PROGRAM VERIFICATION

Predicted values of unstabilized ship roll motion using the one-degree-of-freedom roll motion procedure, have been compared to model and full-scale experiment results in both References 1 and 2. Ongoing work at DTNSRDC by Meyers has found the results of the single-degree equation very similar to those of the coupled, three-degree equations for roll-sway-yaw for the worst heading roll motion, although some underprediction in bow seas and some overprediction in following to quartering seas have been noted. Additionally, Reference 2, as well as work by Lloyd and other Admiralty Marine Technology Establishment (AMTE) personnel, has substantiated, at least in part, the stabilized roll and fin predictions. It is generally recognized that the predictions of FINCON (e.g., for the worst heading) are appropriate for use in design problems.

The coding of FINCON has been verified by making comparisons with results of the older unpublished FINS program, as well as with published results of programs currently used by Lloyd and others at AMTE. The comparisons substantiate the correctness of the coding of FINCON in general, though some differences do occur between the results of FINCON and the

AMTE program. For example, at some speeds, the FINCON unstabilized roll angles were higher than the corresponding AMTE values for the same narrow-beam LEANDER-class frigate evaluated for the case when roll damping is independent of roll angle. The differences may be due to differences in the input, slight differences in the motion equations and algorithms programmed, or differences in the short-crested seas algorithm. One known difference is that FINCON accepts as input the true value for BM, the center of buoyancy; whereas the AMTE program computes a value based on the shape of the waterplane. Another difference, though not relevant to the comparisons made for the LEANDER, is that the AMTE program has no provision to handle the case when roll damping is dependent on roll angle.

FUTURE WORK

FINCON is the first of a series of new tools being developed to enhance the U.S. Navy's roll/fin design capability. As such, FINCON is the basis for all such current and near-future investigations. Specific guidelines for optimum use of the program in the form of a rather complete design exercise can be found in Reference 1. Procedures for evaluating bilge-keel and fin sizing and stabilizer control optimization are detailed there. These procedures indicate how the use of the FINCON program in roll/fin/controller design practice can be extremely instructive.

Another very important area currently being investigated by Cox is the use of a coupled, three-degree system of equations for roll, sway, and yaw motion prediction. A more general and refined program is being developed in conjunction with that work. The program under development will also be of practical use at an early stage of ship design, requiring only very simple input requirements. A complete report and user's manual for the improved three-degree-of-freedom simulation system will soon be published. It will include details of the approach which is used in the current one-degree-of-freedom program to recognize fin-saturation effects.

CONCLUDING REMARKS

This report provides a user's guide to FINCON, a roll, fin, fin controller prediction computer program. No attempt to describe design

practices or the required engineering decisions necessary to using this tool has been made; Reference 1 provides a comprehensive discussion of such materials. Sample inputs and outputs, as well as a description of the program organization and procedures have been given. It is envisioned that the engineer, with a working knowledge of Reference 1, will run the program essentially as a "black box"--he/she is not expected to need to contend with the actual FORTRAN or source deck; and, thus, only a very rudimentary knowledge of programming or computers is required. Instead, he/she will be required to actively participate in the engineering tradeoff decisions necessary in design work, and, as such will probably run FINCON several times in any given investigation.

APPENDIX A

PROGRAM PROCEDURE AND FLOW

The equations solved in FINCON are listed in Table 9 and are taken almost exclusively from Reference 1. For purposes of illustration, a short-crested spreading angle of 10 degrees is assumed. Table 10 provides a description of the nomenclature used in Table 9. The corresponding FORTRAN notation (e.g., see the listing of Appendix B) follows as closely as possible that of Table 10.*

Table 9 is, in a sense, broken into three algorithmic steps. Four basic predictions are identified: stabilized roll, roll rate, fin angle, and fin velocity. Each of the four is identified by a more or less reverse building-block procedure. For example, the final step of the first algorithm is labeled 1-1, the step preceding 1-1 is 1-2, the step preceding 1-2 is 1-3, etc. Similarly, steps 2-1, 2-2, etc., and 3-1, 3-2, etc., are developed. It is felt that this reverse building-block approach to listing the steps makes it easier to see the final results and is also representative of the procedure followed in organizing the equations of Reference 1 for programming purposes. Obviously some of the steps developed for the first algorithm are needed by the other two algorithms (e.g., step 1-4-1); however, it was not considered necessary to repeat these for each of the other two. Instead, for clarity, one can assume that the results of each step of the first algorithm are available to the remaining algorithms.

Figure 3 presents a diagram of the flow sequence of FINCON. The figure identifies the important loops over sea conditions and ship speed for both unstabilized and stabilized predictions. The diagram was constructed with the intent of providing a quick overview of the entire program flow so that major computational segments are easily identified.

*One exception to this is that c_u in Table 9 becomes CA in the FORTRAN.

TABLE 9 - EQUATIONS FOR ROLL, FIN ANGLE, AND
FIN VELOCITY CALCULATION

$$\underline{1-1} \quad [\sigma_{\phi_s}(\mu)]_{sc}^2 = \frac{1}{9} \sum_{p=-8}^8 \cos^2 \left(\frac{p\pi}{18} \right) \left[\sigma_s \left(\mu + \frac{p\pi}{18} \right) \right]_{cn}^2$$

$$\underline{2-1} \quad [\sigma_{\phi}^*(\mu)]_{sc}^2 = \frac{1}{9} \sum_{p=-8}^8 \cos^2 \left(\frac{p\pi}{18} \right) \left[\sigma_{\phi}^* \left(\mu + \frac{p\pi}{18} \right) \right]_{cn}^2$$

$$\underline{3-1} \quad [\sigma_{\beta}(\mu)]_{sc}^2 = \frac{1}{9} \sum_{p=-8}^8 \cos^2 \left(\frac{p\pi}{18} \right) \left[\sigma_{\beta} \left(\mu + \frac{p\pi}{18} \right) \right]_{cn}^2$$

$$\underline{4-1} \quad [\sigma_{\beta}^*(\mu)]_{sc}^2 = \frac{1}{9} \sum_{p=-8}^8 \cos^2 \left(\frac{p\pi}{18} \right) \left[\sigma_{\beta}^* \left(\mu + \frac{p\pi}{18} \right) \right]_{cn}^2$$

$$\underline{1-2} \quad [\sigma_s(v)]_{cn}^2 = \int_0^{\omega^*} S_{\phi_u} [\omega, \omega_E(\omega), v, n_u] \left(\frac{\phi_s}{\phi_u} \right)^2 d\omega$$

$$\underline{2-2} \quad [\sigma_{\phi}^*(v)]_{cn}^2 = \int_0^{\omega^*} S_{\phi_u}^* [\omega, \omega_E(\omega), v, n_u] \left(\frac{\phi_s}{\phi_u} \right)^2 d\omega$$

$$\underline{3-2} \quad [\sigma_{\beta}(v)]_{cn}^2 = \int_0^{\omega^*} S_{\phi_u} [\omega, \omega_E(\omega), v, n_u] \left(\frac{\phi_s}{\phi_u} \right)^2 \frac{[(\beta_a)_o / \phi_s]^2}{a_R^2 + a_I^2} d\omega$$

$$\underline{4-2} \quad [\sigma_{\beta}^*(v)]_{cn}^2 = \int_0^{\omega^*} S_{\phi_u} [\omega, \omega_E(\omega), v, n_u] \left(\frac{\phi_s}{\phi_u} \right)^2 \frac{[(\beta_a)_o / \phi_s]^2}{a_R^2 + a_I^2} \omega_E^2 d\omega$$

TABLE 9 (Continued)

$$\underline{1-3-1} \quad S_{\phi_u} [\omega, \omega_E(\omega), \nu, n_u] = S_{\phi_u} [\omega, \omega_E(\omega), \nu, n_u] (\omega_E(\omega))^2$$

$$\underline{1-3-2} \quad S_{\phi_u} [\omega, \omega_E(\omega), \nu, n_u] = S_{\alpha}(\omega) \left[T_{\phi_u}(\omega_E, \nu, n_u) \right]^2$$

$$\underline{1-3-3} \quad S_{\alpha}(\omega) = \left\{ \frac{487.0626}{T_o^4 \omega^5} (\tilde{\zeta}_w)^2 \exp \left[\frac{-1948.2444}{T_o^4 \omega^4} \right] \right\} \left(\frac{360 \omega^2}{2\pi g} \right)$$

$$\underline{1-3-4} \quad T_{\phi_u}(\omega_E, \nu, n_u) = \frac{\phi_u}{k_{\zeta_a}} = \frac{e^{-kT}}{c_u} \sin \nu (h^2 + C^2 b_u^2)^{1/2}$$

$$\underline{1-3-5} \quad k = 2\pi/\lambda = \omega^2/g; \quad c_u = (a^2 + b_u^2)^{1/2}; \quad a = 1 - \Lambda^2;$$

$$\Lambda = \omega_E/\omega_{\phi}; \quad b_u = 2n_u \Lambda; \quad h = D - qC\Lambda^2;$$

$$D = \frac{\sin k_L}{k_L} \text{ or } F_p + \frac{BM}{GM} G_p \text{ or } F_e + \frac{BM}{GM} G_e \text{ for}$$

rectangular, parabolic, or elliptical waterplanes, respectively;

$$k_L = \frac{1}{2} kL \cos \nu = \frac{1}{2g} \omega^2 L \cos \nu;$$

$$F_p = \frac{3}{k_L^3} [\sin k_L - k_L \cos k_L];$$

$$G_p = \frac{1575}{k_L^7} \left[\left(1 - \frac{2k_L^2}{5} \right) \sin k_L - \left(k_L - \frac{k_L^3}{15} \right) \cos k_L \right] - F_p;$$

TABLE 9 (Continued)

$$F_e = \frac{2}{k_L} J_1(k_L);$$

$$G_e = \frac{8}{k_L^2} [F_e - J_0(k_L)] - F_e;$$

$$C = \frac{\sin k_L^*}{k_L^*}; \quad k_L^* = \frac{1}{2} kL^* \cos \nu;$$

$L^* = L, 1/2 L, \text{ or } \sqrt{7}/4 L$ for rectangular, parabolic, or elliptical waterplane, respectively;

$$\underline{1-4-1} \quad \left(\frac{\phi_s}{\phi_u} \right)^2 = \left(\frac{c_u}{c_s} \right)^2 \left[1 + 2 \left(\frac{s_a}{c_s \phi_s} \right) \left(\frac{a}{c_s} \cos \xi + \frac{b_s}{c_s} \sin \xi \right) + \left(\frac{s_a}{c_s \phi_s} \right)^2 \right]^{-1}$$

$$\underline{1-4-2} \quad c_s = (a^2 + b_s^2)^{1/2}; \quad b_s = 2n_s \Lambda$$

$$\underline{1-4-3} \quad \frac{s_a}{c_s \phi_s} = S_{sm} \frac{(\rho V^2)}{\Delta GM} \text{MAR} \left(\frac{dC_L}{d\beta} \right) E \frac{1}{c_s (a_R^2 + a_I^2)^{1/2}} \frac{(\beta_a)_o}{\phi_s}$$

$$\underline{1-4-4} \quad \sin \xi = \frac{k_I (a_R b_R - a_I b_I) - k_R (a_R b_I + a_I b_R)}{[(k_R^2 + k_I^2) (a_R^2 + a_I^2) (b_R^2 + b_I^2)]^{1/2}};$$

$$\cos \xi = \frac{k_R (a_R b_R - a_I b_I) + k_I (a_R b_I + a_I b_R)}{[(k_R^2 + k_I^2) (a_R^2 + a_I^2) (b_R^2 + b_I^2)]^{1/2}}$$

TABLE 9 (Continued)

$$\underline{1-4-5} \quad \frac{(\beta_a)_0}{\phi_s} = G_k \cdot G_V \left[\frac{k_R^2 + k_I^2}{b_R^2 + b_I^2} \right]^{1/2};$$

$$k_R = k_1 - \omega_E^2 k_3; \quad k_I = \omega_E k_2; \quad b_R = b_1 - \omega_E^2 b_3;$$

$$b_I = \omega_E b_2; \quad a_R = a_1 - \omega_E^2 a_3; \quad a_I = \omega_E a_2;$$

$$\left(\frac{dC_L}{d\beta} \right)_E = h_F(\omega_E) \left(\frac{dC_L}{d\beta} \right)_{FS}; \quad h_F(\omega_E) = h_0 + h_1 \omega_E + h_2 \omega_E^2 \\ + h_3 \omega_E^3 + h_4 \omega_E^4;$$

$$\omega_E = \omega - \frac{\omega^2}{g} v \cos v$$

$$\underline{1-5-1} \quad S_{sm} = \frac{1}{(1-x_2)} \left[(1-Fx_2) \operatorname{erf} \left(\frac{\beta_{stp}}{\sigma_\beta \sqrt{2}} \right) + (F-1)x_2 \operatorname{erf} \left(\frac{\beta_{stp}}{x_2 \sigma_\beta \sqrt{2}} \right) \right]$$

$$\underline{1-5-2} \quad F = \left(\frac{4}{\pi} \right) \left(\frac{x_2}{x_1} \right)$$

$$\underline{1-5-3} \quad x_1 = \left(\frac{\beta_{stp}}{\beta_{max}} \right) \left(\frac{\sigma_\beta^*}{\sigma_\beta} \right)$$

TABLE 10 - NOTATION USED IN EQUATIONS OF TABLE 9

A	Fin area
a	$1 - \Lambda^2$
a_I	$\omega_E a_2$
a_R	$a_I - \omega_E^2 a_3$
a_1, a_2, a_3	Fin servo coefficients
BM	Distance between center of buoyancy and metacenter
b_I	$\omega_E b_2$
b_R	$b_1 - \omega_E^2 b_3$
b_1, b_2, b_3	Fin controller compensation coefficients
$c_{u,s}$	$(a^2 + b_{u,s}^2)^{1/2}$
$(dC_L/d\beta)_E$	Effective lift curve slope
$(dC_L/d\beta)_{FS}$	Free stream lift coefficient curve slope
GM	Transverse metacentric height
G_k	Fin controller overall gain control
G_V	Fin controller velocity dependent gain control
g	Acceleration due to gravity
$h_F(\omega_E)$	Ratio of effective to free stream fin lift curve slopes
h_0, h_1, h_2, h_3, h_4	$h_F(\omega_E) = h_0 + h_1 \omega_E + h_2 \omega_E^2 + h_3 \omega_E^3 + h_4 \omega_E^4$
J_0, J_1	Bessel functions of the first kind
k	Wave number, ω^2/g
k_I	$\omega_E k_2$
k_R	$k_1 - \omega_E^2 k_3$

TABLE 10 (Continued)

k_1, k_2, k_3	Roll angle, velocity, and acceleration characteristic gain values
L	Ship length between perpendiculars
M	Number of fin pairs
n_u	Roll decay coefficient (ship without fins)
q	$\delta I / (I + \delta I)$, see Equation (5) for definition of $\delta I / I$
R	Fin moment arm (about a longitudinal line through the ship center of gravity)
S_{sm}	Saturation multiplier
S_α	Wave-slope spectral coordinate
S_{ϕ_u}	Unstabilized roll response
T	Mean ship draft
T_o	Modal (peak) period of wave-height spectrum
T_{ϕ_u}	Unstabilized roll transfer function
V	Ship speed
$\beta(\beta_a)$	Fin angle (amplitude)
β_{stp}	Fin limit angle
$\dot{\beta}_{max}$	Fin limit velocity
Δ	Ship displacement weight
ζ_a	Wave amplitude
$(\zeta_w^y)^{1/3}$	Significant wave height
Λ	Tuning factor, ω_E / ω_ϕ
λ	Wavelength
μ	Ship heading, with respect to the ship, of predominant wave direction

TABLE 10 (Continued)

ν	Wave direction, with respect to the ship, apart from the predominant direction
ρ	Density of seawater
σ_s	Stabilized root mean square roll angle
σ_β	Root mean square fin angle
σ_β^2	Root mean square fin velocity
ϕ_s	Stabilized roll angle amplitude
ϕ_u	Unstabilized roll angle amplitude
ϕ_s/ϕ_u	Roll reduction factor
ω	Wave frequency
ω_E	Frequency of wave encounter
ω_ϕ	Ship natural roll frequency
ω^*	Frequency above which roll response is negligible, 2 radians/second

Subscripts

lcn	Long-crested
sc	Short-crested
s	Stabilized
u	Unstabilized

FINCON PROGRAM FLOW

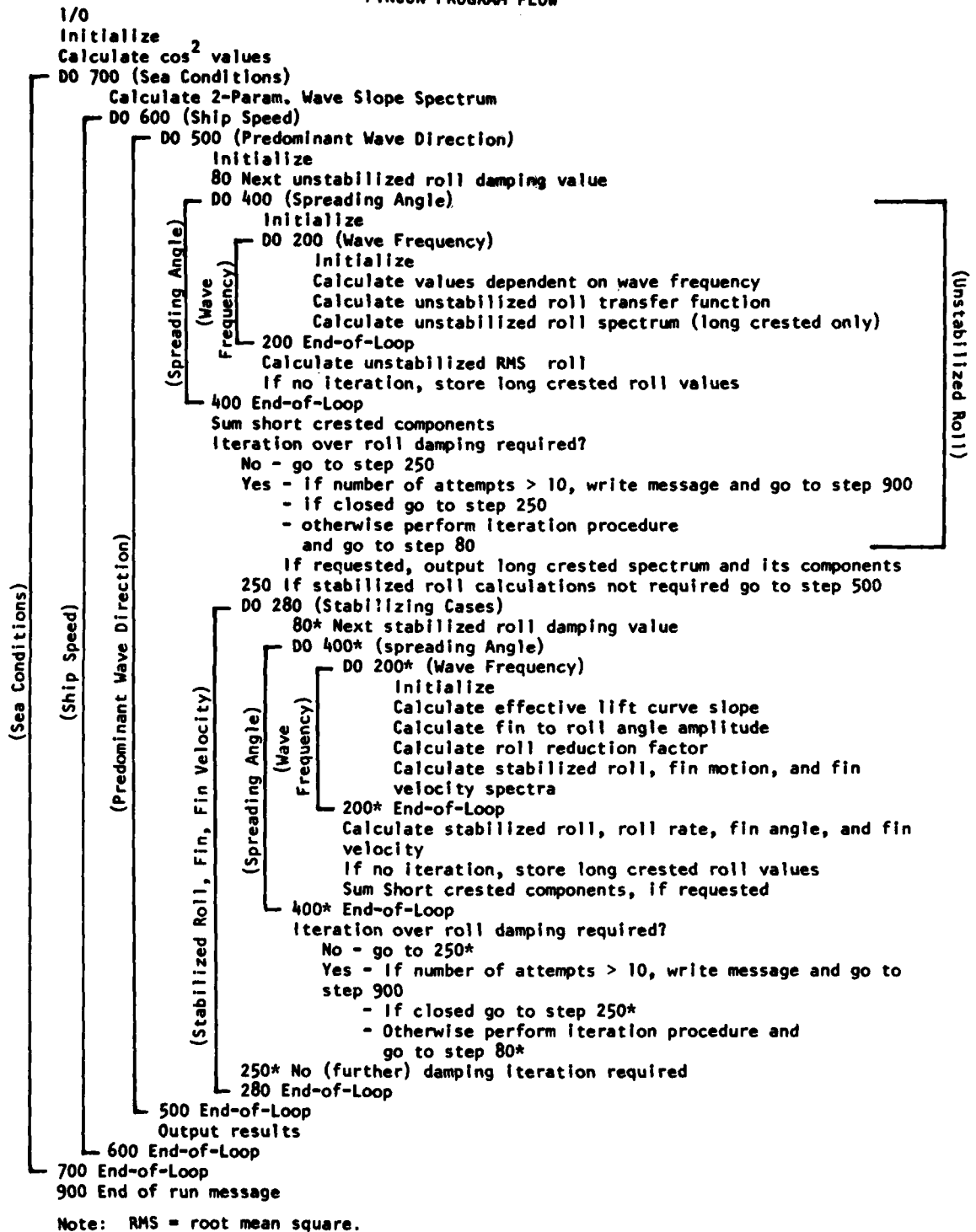


Figure 3 - Diagram of Program Flow Sequence

APPENDIX B

PROGRAM LISTING

A complete listing of FINCON is given on the subsequent pages. The routines are listed in the order FINCON, ITREQ, BRWSSP, ALGRNG, FINSTAB, and ERRF. BESSJ is a system routine and hence not given here. The listing, made from an UPDATE file stored on a private disk pack, contains a unique identification of each line of coding on the far right-hand side of the pages. This identification first identifies the routine by a name (e.g., ROLL, TREQ, BWSS, ALGR, and FST) and then by a line number.

The FORTRAN is embedded with comment cards for identification of the steps of the prediction procedure. Modifications can easily be made to this source coding by inserting or deleting statements anywhere in the routines via an "UPDATE" run of the program.

```

1      PROGRAM FINCON (INPUT=512,OUTPUT=512,TAPE5=INPUT,TAPE6=OUTPUT,
2      TAPE1)
3      ROLL 2
4      ROLL 3
5      ROLL 4
6      ROLL 5
7      ROLL 6
8      ROLL 7
9      ROLL 8
10     ROLL 9
11     ROLL 10
12     ROLL 11
13     ROLL 12
14     ROLL 13
15     ROLL 14
16     ROLL 15
17     ROLL 16
18     ROLL 17
19     ROLL 18
20     ROLL 19
21     ROLL 20
22     ROLL 21
23     ROLL 22
24     ROLL 23
25     ROLL 24
26     ROLL 25
27     ROLL 26
28     ROLL 27
29     ROLL 28
30     ROLL 29
31     ROLL 30
32     ROLL 31
33     ROLL 32
34     ROLL 33
35     ROLL 34
36     ROLL 35
37     ROLL 36
38     ROLL 37
39     ROLL 38
40     ROLL 39
41     ROLL 40
42     ROLL 41
43     ROLL 42
44     ROLL 43
45     ROLL 44
46     ROLL 45
47     ROLL 46
48     ROLL 47
49     ROLL 48
50     ROLL 49
51     ROLL 50
52     ROLL 51
53     ROLL 52
54     ROLL 53
55     ROLL 54
56     ROLL 55
57     ROLL 56
58     ROLL 57
59     ROLL 58

```

*CNC 6700 - - - MARCH, 1976 - - - OTNSRDC - - - CODE 1568, SUSAN SALES
 * J. R. TUCKER, - - AUGUST, 1979 - - CHI ASSOCIATES, INC., ROSLYN, VA.
 *FORTRAN PROGRAM TO PREDICT ROLL MOTIONUNSTABILIZED MOTION PREDICTE
 *USING THE THEORY OF J. E. CONOLLY MODIFIED TO ALLOW FOR NONLINEAR
 *DAMPING. STABILIZED MOTION PREDICTED USING COX AND LLOYD.
 *
 C THIS VERSION OF FINCON (TPSC78) WRITES TO PF THE SHORTCRESTED STAB.
 C OR UNSTAB. ROLL RESPONSES (RMS). THIS FILE IS DESIGNED TO BE POST
 C PROCESSED FOR USE WITH THE POLAR PLOT PROGRAM.
 C
 INTEGER SHAPE,PARAB,ELLIP,RECTANG,SPECTRA
 REAL KL,KLPRIME,L,MUR,MUR,MOUR,LPRIME,LAM,LANCML,K1,K2,K3,M
 COMMON/ITRTN/STAT,PRCN,NTRY,KMU,KNU,KV,GP,GPM1,YPM1,YPP1,
 2 PHIN,IPRINT(8),ITERATE,NPHI
 COMMON /STAB/ NSTAB,M(10),AREA(10),R(10),OCLDBFS(10),H0(10),
 2 H1(10),H2(10),H3(10),H4(10),GK(10),GV(10),K1(10),K2(10),K3(10),
 2 A1(10),A2(10),A3(10),B1(10),B2(10),B3(10),DAMP(10),DAMPS(10,13),
 2 DUC(5,6),SSIGLC(10,13),NNU,ILC,ISC,WE(40,35),
 2 SSIGVLC(10,13),SSIGVSC(10,13),
 2 SUR(40,35),W(40),NM,COSI(35),COM1,SSIGSC(10,13),
 2 VFS(5),DISPLB,GM , BMOTLC(10,13),
 2 BMOTSC(10,13),BVELSC(10,13),BVELL(10,13),BSTOP(5),9VELMAX(5)
 2 ,NSAT,ITEST(10,13)
 DIMENSION TITLE2(8),EKT(40),P(35),S(40),SMW(10),T0(10),VK(5),
 2 MUR(13),MUD(13),NUR(35),SINNU(35),COSNU(35),SIGSQSC(13),SIGSC(13)
 2 ,SIGLC(13),TR(40),MOUR(35),BESSEL(100),SURV(40,35)
 2 ,SGVSC(13),SIGVSC(13),VMOUR(35),SIGVLC(13),SMWET(10)
 DATA SPECTRA,ITERATN/7HSPECTRA,7HITERATN/
 DATA MUD/0.15,30.45,60.75,90.105,120.135,150.165,180/
 DATA PI,RH3,GRAVITY/3.1415926,1.99,32.1725/
 DATA PARA9,ELLIP,RECTANG/10HPARAB ,10HELLIP ,
 210HRECTANG /
 DATA NW/40/ , NNU/13/ ,EPS/.0001/, MDEL/.05/
 *
 *INPUT AND OUTPUT THE SEA AND SHIP CONDITIONS FOR UNSTABILIZED ROLL
 *CALCULATIONS.
 *
 READ (5,1000) NAME1,NAME2,NAME3
 WRITE (6,2000) NAME1,NAME2,NAME3
 READ (5,1000) TITLE2
 WRITE (6,2001) TITLE2
 READ (5,1001) NWH,NV,IUNITS
 WRITE (1) NWH,NV
 READ (5,1002) (SMW(I),I=1,NWH)
 *
 IF (IUNITS.EQ. 0) GO TO 5
 WRITE (6,3002) (SMW(I),I=1,NWH)
 GO TO 6
 *
 5 WRITE (6,2002) (SMW(I),I=1,NWH)
 6 CONTINUE
 READ (5,1002) (T0(I),I=1,NW4)
 WRITE (6,2003) (T0(I),I=1,NWH)
 READ (5,1002) (VK(I),I=1,NV)

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	WRITE (6,2008) (VK(I),I=1,NV)	ROLL	59
	READ (5,1002) DISPTON,L,T,GH,BN,TPHI,Q	ROLL	60
60	* IF (IUNITS .EQ. 0) GO TO 7	ROLL	61
	WRITE (6,3004) DISPTON,L,T,GH,BN,TPHI,Q	ROLL	62
	GO TO 4	ROLL	63
	* 7 WRITE (6,2004) DISPTON,L,T,GH,BN,TPHI,Q	ROLL	64
65	8 CONTINUE	ROLL	65
	READ (5,1000) SHAPE	ROLL	66
	WRITE (6,2005) SHAPE	ROLL	67
	READ (5,1004) ISC, ANGLE	ROLL	68
70	IF (ISC.EQ.0) GO TO 10	ROLL	69
	WRITE (6,2006) ANGLE	ROLL	70
	10 READ (5,1004) ITERATE	ROLL	71
	STAT = 1.0	ROLL	72
	PRGN = 0.01	ROLL	73
75	IF (ITERATE .GT. 0) WRITE (6,2017)	ROLL	74
	IF (ITERATE .EQ.0) READ (5,1002) (DUC(IV,1),IV=1,NV)	ROLL	75
	IF (ITERATE .EQ.0) WRITE (6,2016) (DUC(IV,1),IV=1,NV)	ROLL	76
	IF (ITERATE .EQ.0) GO TO 20	ROLL	77
	WRITE (6,2007)	ROLL	78
80	DO 15 IV=1,NV	ROLL	79
	READ (5,1002) (DUC(IV,I),I=1,6)	ROLL	80
	15 WRITE (6,2009) VK(IV),IDUC(IV,I),I=1,6	ROLL	81
	20 READ (5,1000) IPRINT	ROLL	82
	IF (ISC.NE.0).AND.(IPRINT(1).EQ.SPECTRA) GO TO 22	ROLL	83
85	IF (IPRINT(1) .EQ. SPECTRA) WRITE (6,2018)	ROLL	84
	IF (IPRINT(2) .EQ. ITERATN) WRITE (6,2019)	ROLL	85
	GO TO 25	ROLL	86
	22 WRITE (6,2029)	ROLL	87
	GO TO 900	ROLL	88
90	* *INPUT AND OUTPUT CONDITIONS FOR FIN STABILIZED ROLL PREDICTIONS.	ROLL	89
	* 25 READ (5,1001) NSTAB	ROLL	90
	IF (NSTAB .LT. 1) GO TO 35	ROLL	91
95	IF (NSTAB .GE. 1 .AND. IUNITS .EQ. 0) WRITE (6,2024) NSTAB	ROLL	92
	IF (NSTAB .GE. 1 .AND. IUNITS .EQ. 1) WRITE (6,3024) NSTAB	ROLL	93
	DO 28 I=1,NSTAB	ROLL	94
	READ (5,1004) M(I),AREA(I),R(I)	ROLL	95
	READ (5,1002) DCLOBFS(I),M0(I),M1(I),M2(I),M3(I),M4(I)	ROLL	96
100	READ (5,1002) GK(I),GV(I),K1(I),K2(I),K3(I)	ROLL	97
	READ (5,1002) A1(I),A2(I),A3(I),B1(I),B2(I),B3(I)	ROLL	98
	28 WRITE (6,2025) I,M(I),AREA(I),R(I),DCLOBFS(I),M0(I),M1(I),M2(I),	ROLL	99
	2 M3(I),M4(I),GK(I),GV(I),K1(I),K2(I),K3(I),A1(I),A2(I),A3(I),	ROLL	100
	2 B1(I),B2(I),B3(I)	ROLL	101
105	READ (5,1001) NSAT	ROLL	102
	IF (NSAT.EQ.0) GO TO 30	ROLL	103
	READ (5,1002) (BSTOP(IV),BVELMAX(IV),IV = 1,NV)	ROLL	104
	DO 29 IV = 1,NV	ROLL	105
	29 IF (BVELMAX(IV).EQ.0.0) BVELMAX(IV) = 10.*BSTOP(IV)/TPHI	ROLL	106
110	WRITE (6,2028) (IV,BSTOP(IV),BVELMAX(IV),IV = 1,NV)	ROLL	107
	30 CONTINUE	ROLL	108
	35 CONTINUE	ROLL	109
	* *INITIALIZE.	ROLL	110
		ROLL	111
		ROLL	112
		ROLL	113
		ROLL	114
		ROLL	115

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115	*	IF (IUNITS .EQ. 0) GO TO 30	ROLL	116
	*		ROLL	117
		UM = 3.28054	ROLL	118
		DO 36 I = 1,NMH	ROLL	119
120		SMH(I) = SMH(I)	ROLL	120
		SMH(I) = SMH(I)*UM	ROLL	121
	36	CONTINUE	ROLL	122
		DISPTON = DISPTON*0.9842	ROLL	123
		L = L*UM	ROLL	124
125		T = T*UM	ROLL	125
		GM = GM*UM	ROLL	126
		QM = QM*UM	ROLL	127
	*		ROLL	128
		IF (NSTAB .LT. 1) GO TO 38	ROLL	129
130		DO 37 I = 1,NSTAB	ROLL	130
		AREA(I) = AREA(I)*10.7639	ROLL	131
		R(I) = R(I)*UM	ROLL	132
	37	CONTINUE	ROLL	133
	*		ROLL	134
135		38 CONTINUE	ROLL	135
	*		ROLL	136
		WPHI=2.*PI/TPHI	ROLL	137
		DISPL9 = 2240. * DISPTON	ROLL	138
		DO 40 IV=1,NV	ROLL	139
140		VFS(IV) = 1.6878* VK(IV)	ROLL	140
	40	TER41 = 2.*PI*GRAVITY/L	ROLL	141
		DO 60 IW=1,NM	ROLL	142
		W(IW)=FLOAT(IW)*WDEL	ROLL	143
		60 FKT(IW) = EXP (-W(IW)**2/GRAVITY * T)	ROLL	144
145	*		ROLL	145
		*GENERATE VALUES FOR COSINE SQUARED SPREADING OF WAVE ENERGY.	ROLL	146
	*		ROLL	147
		IF (ISC .EQ. 0) GO TO 75	ROLL	148
		HA = (180./ANGLE)/2.	ROLL	149
150		CON1 = 1./4A	ROLL	150
		NNU = 2*IFIX(HA) - 1	ROLL	151
		ILC = NNU/2 + 1	ROLL	152
		CON2 = PI/(2.*HA)	ROLL	153
		J = -IFIX(HA)	ROLL	154
155		DO 70 I=1,NNU	ROLL	155
		J = J + 1	ROLL	156
		P(I) = J	ROLL	157
	70	COS1(I) = COS(P(I)*CON2)**2	ROLL	158
	75	CONTINUE	ROLL	159
160	*		ROLL	160
		*BEGIN LOOP OVER SEA CONDITIONS.	ROLL	161
	*		ROLL	162
		DO 700 IWM=1,NMH	ROLL	163
	*		ROLL	164
165		*COMPUTE BRETSCHNEIDER 2-PARAMETER WAVE SLOPE SPECTRUM.	ROLL	165
	*		ROLL	166
		CALL BRWSSP (NM,SMH(IWM),T0(IWM),W,S)	ROLL	167
	*		ROLL	168
		*BEGIN LOOP OVER SHIP. SPEED.	ROLL	169
170	*		ROLL	170
		DO 600 IV=1,NV	ROLL	171
			ROLL	172

	* BEGIN LOOP OVER SHIP HEADING (NU), PREDOMINANT WAVE DIRECTION.	ROLL	173
		ROLL	174
175	DO 500 INU=1, NNU	ROLL	175
	IF ((IV.EQ.1) .AND. (IWM.EQ.1))	ROLL	176
	2 MUR(INU) = PI/180. * FLOAT(MUD(INU))	ROLL	177
	DAMPU(INU) = DUC(IV,1)	ROLL	178
	IF (ITEPATE.EQ.0) GO TO 90	ROLL	179
180	NTRY = 0	ROLL	180
	YP = 0.0	ROLL	181
	80 NTRY = NTRY + 1	ROLL	182
	X = YP	ROLL	183
	DAMPU(INU) = DUC(IV,1) + 1.61* DUC(IV,2)*X**0.772 + 1.88* DUC(IV,3)	ROLL	184
185	2)*X + 4.00* DUC(IV,4)*X**2.0 + 9.40* DUC(IV,5)*X**3.0 +	ROLL	185
	2 24.0* DUC(IV,6)*X**4.0	ROLL	186
	90 SIGLVC(INU) = SIGSOSC(INU) = 0.	ROLL	187
	SIGVLC(INU) = SGVSOSC(INU) = 0.	ROLL	188
		ROLL	189
190	* BEGIN LOOP OVER SPREADING ANGLE (NU).	ROLL	190
		ROLL	191
	* (FOR PURELY LONG CRESTED-CASE, NO. OF NU'S = 1, AND NU = 1U.)	ROLL	192
		ROLL	193
	IF (ISC.EQ.0) NNU = 1	ROLL	194
195	DO 400 INU=1, NNU	ROLL	195
	MUR(INU) = MUR(INU) + P(INU)*C0Y2	ROLL	196
	SINMU(INU) = SIN(MUR(INU))	ROLL	197
	50 COSMU(INU) = COS(MUR(INU))	ROLL	198
		ROLL	199
200	* BEGIN LOOP OVER WAVE FREQUENCY.	ROLL	200
		ROLL	201
	DO 200 IW=1, NW	ROLL	202
	KL = L * W(IW)**2 * COSMU(INU) / (2. * GRAVITY)	ROLL	203
	SINKL = SIN(KL)	ROLL	204
205	COSKL = COS(KL)	ROLL	205
	WE(IW, INU) = ABS(W(IW)) * (1. - W(IW) * VFS(IV) / GRAVITY *	ROLL	206
	2 COSMU(INU))	ROLL	207
	TUNF = WE(IW, INU)/WPHI	ROLL	208
	BA = 2. * DAMPU(INU) * TUNF	ROLL	209
210	A = 1. - TJNF * TUNF	ROLL	210
	CA = SORT(A*A + BA*BA)	ROLL	211
		ROLL	212
	* TEST FOR WATERPLANE SHAPE.	ROLL	213
		ROLL	214
215	IF (SHAPE.EQ. ELLIP) GO TO 110	ROLL	215
	IF (SHAPE.EQ. RECTANG) GO TO 120	ROLL	216
		ROLL	217
	* WATERPLANE IS PARABOLIC.	ROLL	218
		ROLL	219
220	LPRIME = L * .5	ROLL	220
	IF (ABS(KL) .GT. EPS) GO TO 105	ROLL	221
	F = 1.	ROLL	222
	G = 0.	ROLL	223
	D = 1.	ROLL	224
225	GO TO 127	ROLL	225
	105 F = 3. * (SINKL - KL * COSKL) / KL**3	ROLL	226
	G = 1575. / KL**7 * ((1. - 2. * KL**2 / 5.) * SINKL	ROLL	227
	2 - KL * (1. - KL**2 / 15.) * COSKL) - F	ROLL	228
		ROLL	229

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230      D = F + BM / GM * G
      GO TO 127
      *
      *WATERPLANE IS ELIPTICAL. (DETERMINE THE ZEROth AND FIRST BESSEL
      *FUNCTIONS OF KL.)
      *
235      110 LPRIME = SORT(7./4. * L
      IF (ABS(KL) .GT. EPS) GO TO 115
      F = 1.
      G = 0.
      D = 1.
240      GO TO 127
      115 CALL BESSJ (KL,D.,1,BESSEL)
      F = 2. / KL * BESSEL(2)
      G = - 9. / KL**2 * BESSEL(1) + (8. / KL**2 - 1.) * F
      D = F + BM / GM * G
245      GO TO 127
      *
      *WATERPLANE IS RECTANGULAR.
      *
250      120 LPRIME = L
      IF (ABS(KL) .GT. EPS) GO TO 125
      D = 1.
      GO TO 127
      125 D = SINKL / KL
      127 CONTINUE
255      KLPRIME = .5 * W(IW)**2 / GRAVITY * LPRIME * (COSNU(INU)
      C = SIN(KLPRIME) / KLPRIME
      H = D - Q*C*TUNF*TUNF
      *
      *COMPUTE ROLL TRANSFER FUNCTION.
260      *
      TRIW = EXT(IW)/CA*SINNU(INU)*SQRT(H*H+C*C*BA*BA)
      *
      *COMPUTE ROLL SPECTRUM.
      *
265      SUR(IW,INU) = SIW*TRIW*TRIW
      SURV(IW,INU) = SUR(IW,INU)*WE(IW,INU)*WE(IW,INU)
      *
      *END OF LOOP OVER WAVE FREQUENCY.
      *
270      200 CONTINUE
      *
      *DETERMINE RMS ROLL VALUE.
      *
275      CALL ALGRNG (NW,N,SUR(1,INU),MOUR(INU))
      CALL ALGRNG (NW,N,SURV(1,INU),VMOUR(INU))
      *
      * IF REQUESTED COMPUTE LONG-CRESTED VALUES
      *
      IF (ISC .NE. 0) GO TO 300
      SIGLC(INU) = SQRT(MOUR(INU))
      SIGVLC(INU) = SQRT(VMOUR(INU))
      *
      * IF REQUESTED, ITERATE OVER ROLL DAMPING FOR LONG-CREST CASE.
      *
285      IF (ITERATE .EQ. 0) GO TO 400

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ROLL 230
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	KWJ = IMU	ROLL	287
	KV = IV	ROLL	288
	KWJ = IMU	ROLL	289
	SGVWLC = SIGVLC(IMU)/WPHI	ROLL	290
290	PHIN = DAMPU(IMU)	ROLL	291
	CALL ITREQ(SGVWLC), RETURNS(00,400,900)	ROLL	292
	* BEGIN SUMMING OF SHORTCRESTED RESPONSE DATA.	ROLL	293
	* 300 SIGSQSC(IMJ) = SIGSQSC(IMU) + COS1(IMU)*HOUR(IMU)	ROLL	294
295	SGVSCSC(IMU) = SGVSCSC(IMU) + COS1(IMU)*HOUR(IMU)	ROLL	295
	* SEND OF LOOP OVER NU.	ROLL	296
	* 400 CONTINUE	ROLL	297
300	* IF (ISC.EQ. 0) IMU = 1	ROLL	298
	IF (ISC.EQ. 0) GO TO 210	ROLL	299
	SIGSC(IMU) = SQRT(CON1*SIGSQSC(IMU))	ROLL	300
305	SIGVSC(IMJ) = SQRT(CON1*SGVSCSC(IMU))	ROLL	301
	SGVWMPH = SIGVSC(IMU)/WPHI	ROLL	302
	KWJ=IMU	ROLL	303
	KWJ=IMU	ROLL	304
	KV=IV	ROLL	305
310	IF (ITERATE.EQ. 0) GO TO 250	ROLL	306
	PHIN=DAMPUI(IMU)	ROLL	307
	CALL ITRE2(SGVWMPH), RETURNS(00,250,900)	ROLL	308
	* IF REQUESTED, PRINT LONGCRESTED SPECTRUM AND ITS COMPONENTS.	ROLL	309
315	* 210 IF (IPRINT(1).NE.SPECTRA) GO TO 250	ROLL	310
	IF (IUNITS.EQ. 0) GO TO 231	ROLL	311
	* WRITE (6,3010) TITLE2,SMNEY(IMH),T0(IMH),VK(IV)	ROLL	312
320	GO TO 232	ROLL	313
	* 231 WRITE (6,2010) TITLE2,SMH(IMH),T0(IMH),VK(IV)	ROLL	314
	CONTINUE	ROLL	315
325	WRITE (6,2023) HUD(IMU)	ROLL	316
	WRITE (6,2020)	ROLL	317
	DO 350 IM=1,NW	ROLL	318
	PER = 2.*PI/MIW	ROLL	319
	LAM = PER*PER*GRAVITY/(2.*PI)	ROLL	320
330	LAMONL = LAM/L	ROLL	321
	WN = 360.*4(IM)*M(IM)/(2.*PI*GRAVITY)	ROLL	322
	RAO=TR(IM)*TR(IM)	ROLL	323
	SD = S(IM)/(WN*WN)	ROLL	324
	RAOD = WN*WN*RAO	ROLL	325
335	SURD = SD * RAOD	ROLL	326
	WRITE (6,2021) MIW,ME(IM,IMU),LAM,LAMONL,WN,TR(IM),RAO,S(IM),	ROLL	327
	2 SUR(IM), RAOD,SD,SURD	ROLL	328
	350 CONTINUE	ROLL	329
	WRITE (6,2022) SIGLC(IMU)	ROLL	330
340	* IF REQUESTED, COMPUTE STABILIZED ROLL AND FIN MOTIONS FOR THIS	ROLL	331
	*PREDOMINANT HEADING.	ROLL	332
		ROLL	333

		ROLL	344
		ROLL	345
345	250 IF (NSTAB .LT. 1) GO TO 500	ROLL	346
	DO 260 IS=1,NSTAB	ROLL	347
	CALL FINSTAB (IS), RETURNS (200,900)	ROLL	348
	260 CONTINUE	ROLL	349
		ROLL	350
	SEND OF LOOP OVER PREDOMINANT HEADING.	ROLL	351
350	500 CONTINUE	ROLL	352
		ROLL	353
	*ONLY LONGCRESTED ROLL VALUES ARE OUTPUT.	ROLL	354
		ROLL	355
355	IF (IUNITS .EQ. 0) GO TO 502	ROLL	356
		ROLL	357
	WRITE (6,3010) TITLE2,SMHET(IWH),T0(IWH),VK(IV)	ROLL	358
	GO TO 503	ROLL	359
		ROLL	360
360	502 WRITE(6,2010) TITLE2,SMH(IWH),T0(IWH),VK(IV)	ROLL	361
	503 CONTINUE	ROLL	362
	WRITE(1) TITLE2,SMH(IWH),T0(IWH),VK(IWH)	ROLL	363
	IF (ISC .NE. 0) GO TO 550	ROLL	364
	IF (ITERATE .EQ. 0) WRITE(6,2030)	ROLL	365
365	WRITE(6,2031)	ROLL	366
	DO 505 IMU = 1,MMU	ROLL	367
	505 WRITE(6,2014) MUD(IMU),DAMPU(IMU),SIGLC(IMU)	ROLL	368
	WRITE(1) (SIGLC(IMU),IMU = 1,MMU)	ROLL	369
	IF (NSTAB .EQ. 0) GO TO 600	ROLL	370
370	DO 540 I = 1,NSTAB	ROLL	371
	WRITE(6,2012) I	ROLL	372
	DO 525 IMU = 1,MMU	ROLL	373
	525 WRITE(6,2014) MUD(IMU),DAMPS(I,IMU),SSIGLC(I,IMU),ITECT(I,IMU),	ROLL	374
	2 BMTLC(I,IMU),BVELLC(I,IMU)	ROLL	375
375	WRITE(1) (SSIGLC(I,IMU), IMU = 1,MMU)	ROLL	376
	540 CONTINUE	ROLL	377
	IF (NSAT .EQ. 0) GO TO 600	ROLL	378
	WRITE(6,2222)	ROLL	379
	GO TO 600	ROLL	380
380		ROLL	381
	*ONLY SHORTCRESTED ROLL VALUES ARE OUTPUT	ROLL	382
		ROLL	383
	550 IF(ITERATE .EQ. 0) WRITE(6,2030)	ROLL	384
	WRITE(6,2013)	ROLL	385
385	DO 555 IMU = 1,MMU	ROLL	386
	555 WRITE(6,2014) MUD(IMU),DAMPU(IMU),SIGSC(IMU)	ROLL	387
	WRITE(1) (SIGSC(IMU),IMU = 1,MMU)	ROLL	388
	IF (NSTAB .EQ. 0) GO TO 600	ROLL	389
	DO 590 I = 1,NSTAB	ROLL	390
390	WRITE(6,2026) I	ROLL	391
	DO 575 IMU = 1,MMU	ROLL	392
	575 WRITE(6,2014) MUD(IMU),DAMPS(I,IMU),SSIGSC(I,IMU),ITECT(I,IMU),	ROLL	393
	2 BMTSC(I,IMU),BVELSC(I,IMU)	ROLL	394
	WRITE(1) (SSIGSC(I,IMU),IMU = 1,MMU)	ROLL	395
395	590 CONTINUE	ROLL	396
	IF (NSAT .EQ. 0) GO TO 600	ROLL	397
	WRITE(6,2222)	ROLL	398
		ROLL	399
	SEND OF LOOP OVER SPEED.	ROLL	400

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400      S      600 CONTINUE                                ROLL      401
      .                                ROLL      402
      SEND OF LOOP OVER SEA CONDITION.                    ROLL      403
      .                                ROLL      404
      700 CONTINUE                                         ROLL      405
      900 WRITE (6,2015)                                   ROLL      406
      .                                ROLL      407
      STOP                                                 ROLL      408
      1000 FORMAT (8A10)                                    ROLL      409
410      1001 FORMAT (16I5)                                ROLL      410
      1002 FORMAT (8F10.6)                                ROLL      411
      1004 FORMAT (I5,2F10.5,A10)                         ROLL      412
      2000 FORMAT (11M1,27(//),45X,42N* * * ROLL MOTION PREDICTION PROGRAM * * ROLL      413
      2 //,56X,3A10)                                       ROLL      414
415      2001 FORMAT (11M1,8A10//)                         ROLL      415
      2002 FORMAT (1X,*SIGNIFICANT WAVE HEIGHT(S) (FEET) =*,5X,10F7.2) ROLL      416
      2003 FORMAT (1X,*MODAL WAVE PERIOD(S) (SECONDS) =*,8X,10F7.2//) ROLL      417
      2004 FORMAT(1X,*DISPLACEMENT (L. TONS) =*16X,F7.8/1X *LENGTH BETWEEN ROLL      418
      2PP (FEET) =*14X,F7.1/1X,*DRAFT (FEET) =* 26X,F7.2/1X, ROLL      419
420      2*TRANSVERSE METACENTRIC HEIGHT (FEET) =*2X,F7.1/1X*METACENTER ABOVE ROLL      420
      2E BUOYANCY CENTER (FT) =*,1X,F7.2/1X,*ROLL PERIOD (SECONDS) =*,17X ROLL      421
      2,F7.2/1X, *Q =*,37X,F7.3) ROLL      422
      2005 FORMAT (1X,*WATERPLANE SHAPE =*,24X,A10) ROLL      423
      2006 FORMAT (1X,*SPREADING ANGLE =*23X,F7.0//) ROLL      424
425      2007 FORMAT(1X,110NDAMPING INPUT IN THE FORM N = C1 + 1.61*C2*Y**0.7 ROLL      425
      272 + 1.88*C3*Y + 4.00*C4*Y**2 + 9.48*C5*Y**3 + 24.88*C6*Y**4// ROLL      426
      2* SPEED (KNOTS)*,4X,*C1*,7X,*C2*,7X,*C3*,7X,*C4*,7X,*C5*,7X,*C6*) ROLL      427
      2008 FORMAT (1X,*SHIP SPEED(S) (KNOTS) =*,19X,9F5.1/) ROLL      428
      2009 FORMAT(4X,F5.1,2X,6(2X,F7.6)) ROLL      429
430      2010 FORMAT (11M1,8A10//1X,*SIGNIFICANT WAVE HEIGHT =*F7.2,* FEET* ROLL      430
      2 / 1X *MODA ROLL      431
      2L WAVE PERIOD =*F7.2* SECONDS*/1X*SHIP SPEED =* F5.1,* KNOTS*) ROLL      432
      2011 FORMAT (1//1X,*UNSTABILIZED R4S ROLL (DEGREES)*/1X,*HEADING*7X*LC* ROLL      433
      2 8X,*SC*/) ROLL      434
435      2012 FORMAT (I6,1X,2F10.2,2X,A1,15X,2F10.2,12X,2F10.2) ROLL      435
      2013 FORMAT (1//1X,*UNSTABILIZED R4S ROLL (DEGREES)*/ 1X,*HEADING* 8X, ROLL      436
      2 *N*8X*SC*/) ROLL      437
      2014 FORMAT (I6,1X,F10.4,F10.2,2X,A1,25X,F10.2,22X,F10.2) ROLL      438
      2015 FORMAT (11M1,27(//),57X,17N* * * E N D * * *) ROLL      439
440      2016 FORMAT (1X,*ROLL DAMPING COEFFICIENT(S) =*,13X,5F5.3//) ROLL      440
      2017 FORMAT (1X,*ITERATION OVER ROLL DAMPING WILL BE DONE.*/) ROLL      441
      2018 FORMAT (1//1X,*LONGCRESTED SPECTRA AND COMPONENTS WILL BE PRINTED. ROLL      442
      2*) ROLL      443
      2019 FORMAT (1//1X,*INTERMEDIATE STEPS IN ROLL DAMPING ITERATION WILL B ROLL      444
      2E PRINTED.*/) ROLL      445
445      2020 FORMAT (1//9X,*W*,8X,*ME*,7X,*LAM*,5X, *LAM/L*,9X,*K*,8X, ROLL      446
      2 *TR*,7X,*RAO*,4X,*M SL S*,7X,*SUR*, 3X,*RAO DIM*,4X,*M HT S*,3X, ROLL      447
      2 *SUR DIM*/ 3X,*RAD/SEC*3X *RAD/SEC*8X,*FT*,17X,*DEG*,3X,*DEG/DEG ROLL      448
      2 DEG/DEG SQ DEGSQ SEC*1X,*DEGSQ SEC*,1X,*DEG/FT SQ*1X,*FTSQ SEC*1X ROLL      449
450      2 *DEGSQ SEC*) ROLL      450
      2021 FORMAT (12F10.3) ROLL      451
      2022 FORMAT (1//1X,*RMS ROLL =*F7.2* DEGREES*) ROLL      452
      2023 FORMAT (1//1X,*SHIP HEADING =*I3* DEGREES*) ROLL      453
      2024 FORMAT (1//1X,*ROLL STABILIZATION WILL BE CALCULATED FOR *I3 ROLL      454
455      2 * CASES*/ 1X *FIN AND CONTROL SYSTEM PARAMETERS ARE AS FOLLOWS:* ROLL      455
      2 // CASE* 2X *N* 6X *A* 5X *R (DCL/DB)FS* 4X *W8* 4X *H1* 4X ROLL      456
      ROLL      457
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PROGRAM FINCON 74/74 OPT=0 ROUND=0/ TRACE FTM 4.6+460 11/26/79 11.25.30

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2 *M2* 4X *M3* 4X *M4* 4X *GK* 4X *GV* 4X *K1* 4X *K2* 4X *K3* 4X ROLL 458
2 *A1* 4X *A2* 4X *A3* 4X *B1* 4X *B2* 4X *B3* /10X *FT SQ* 4X *FT* ROLL 459
2 4X *PER DEG*/) ROLL 460
460 2025 FORMAT (I5,I3,F7.2,F6.2,F12.3,16F6.3) ROLL 461
2026 FORMAT (//1X,*CASE*I3,* STABILIZED RMS ROLL (DEGREES)* ROLL 462
2 10X,*FIN MOTION (DEGREES)* 10X *FIN VELOCITY (DEGREES/SECOND)* ROLL 463
2 1X,*HEADING* 8X *N* 8X *SC* 36X*SC* 30X *SC*/) ROLL 464
2027 FORMAT (//1X,*CASE*I3,* STABILIZED RMS ROLL (DEGREES)* ROLL 465
2 10X,*FIN MOTION (DEGREES)* 10X *FIN VELOCITY (DEGREES/SECOND)* ROLL 466
2 1X,*HEADING* 7X *LC* 8X *SC* 27X *LC* 4X *SC* 19X *LC* 8X *SC*/) ROLL 467
2028 FORMAT (//5X,*IV*5X,*BSTOP*5X,*BVELMAX*/) ROLL 468
2 (I7,2X,F10.5,2X,F10.5,/) ROLL 469
470 2029 FORMAT (//1X,*NO SPECTRA OUTPUT ALLOWED IN SHORTCRESTED CASE. REV ROLL 470
2ISE YOUR INPUT DATA. ROLL 471
2030 FORMAT (//1X,*ITERATION OVER ROLL-DAMPING NOT PERFORMED. ROLL 472
2031 FORMAT (//1X,*UNSTABILIZED RMS ROLL (DEGREES)*/1X,*HEADING*,8X, ROLL 473
2 *N*,8X,*LC*/) ROLL 474
475 2032 FORMAT (//1X,*CASE*I3,* STABILIZED RMS ROLL (DEGREES)* ROLL 475
2 10X,*FIN MOTION (DEGREES)*,10X,*FIN VELOCITY (DEGREES/SECOND)* ROLL 476
2 1X,*HEADING* 8X *N* 8X *LC* 36X *LC* 30X *LC*/) ROLL 477
2222 FORMAT (//1X,36H* SATURATION FACTOR IS INSIGNIFICANT) ROLL 478
3002 FORMAT (1X,*SIGNIFICANT WAVE HEIGHT(S) (METERS) =*,3X,10F7.2) ROLL 479
480 3004 FORMAT (//1X,*DISPLACEMENT (M. TONS) =*23X,F7.0,1X *LENGTH BETWEEN ROLL 480
2PP (METERS) =*14X,F7.1,1X,*DRAFT (METERS) =* 26X,F7.2/1X, ROLL 481
2*TRANSVERSE METACENTRIC HEIGHT (METERS) =*2X,F7.1/1X*METACENTER AB ROLL 482
20VE BUOYANCY CENTER (M) =*,4X,F7.2/1X,*ROLL PERIOD (SECONDS) =*, ROLL 483
219X,F7.2/1X, *Q =*,39X,F7.3) ROLL 484
485 3010 FORMAT (1M1,8A10//1X,*SIGNIFICANT WAVE HEIGHT =*F7.2,* METERS* ROLL 485
2 / 1X *MODA ROLL 486
2L WAVE PERIOD =*F7.2* SECONDS*/1X*SHIP SPEED =* F5.1,* KNOTS*) ROLL 487
3024 FORMAT (//1X,*ROLL STABILIZATION WILL BE CALCULATED FOR *I3 ROLL 488
2 * CASES*/ 1X *FIN AND CONTROL SYSTEM PARAMETERS ARE AS FOLLOWS* ROLL 489
2 /* CASE* 2X *M* 4X *A* 7X *R (DCL/OB)FS* 4X *M0* 4X *M1* 4X ROLL 490
2 *M2* 4X *M3* 4X *M4* 4X *GK* 4X *GV* 4X *K1* 4X *K2* 4X *K3* 4X ROLL 491
2 *A1* 4X *A2* 4X *A3* 4X *B1* 4X *B2* 4X *B3* /11X *M SQ* 5X *M * ROLL 492
2 3X *PER DEG*/) ROLL 493
END ROLL 494

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SUBROUTINE ITREQ

74/74

OPT=0 ROUND=0/ TRACE

FTN 4.6+468

11/26/79 11.25.38

1	SUBROUTINE ITREQ(SIG), RETURNS(A,B,C)	TREQ	2
	COMMON/ITRTN/STAT,PRCN,NTRY,IMU,INU,IV,GP,GPM1,VPM1,YP,VPP1,	TREQ	3
2	PHIN,IPRINT(8),ITERATE,NPMI	TREQ	4
	DATA ITERATN/7HITERATN/	TREQ	5
5	IF (NTRY,37,18) WRITE(6,4998)	TREQ	6
	IF (NTRY,67,18) RETURN C	TREQ	7
	IF ((IPRINT(2) .EQ. ITERATN) .AND. (IMU .EQ. 1) .AND. (NTRY.EQ.1))	TREQ	8
	2 WRITE(6,8101)	TREQ	9
	GP = STAT*SIG	TREQ	10
10	IF (IPRINT(2) .EQ. ITERATN) WRITE(6,8100)IV,IMU,NTRY,PHIN,VPP1,	TREQ	11
	2 YP,VPM1,GP,GPM1	TREQ	12
	IF (GP .LT. .81) RETURN B	TREQ	13
	IF (NTRY .EQ. 1) GO TO 28	TREQ	14
	IF (ABS(1 - GP/YP) .LE. PRCN) RETURN B	TREQ	15
15	VPP1 = (GPM1*YP - GP*VPM1)/((YP - VPM1) - (GP - GPM1))	TREQ	16
	VPM1 = YP	TREQ	17
	GP = GP	TREQ	18
	YP = VPP1	TREQ	19
	IF (IPRINT(2) .EQ. ITERATN) WRITE(6,8100)IV,IMU,NTRY,PHIN,VPP1,	TREQ	20
20	2 YP,VPM1,GP,GPM1	TREQ	21
	RETURN A	TREQ	22
	20 GPM1 = GP	TREQ	23
	VPM1 = 0.)	TREQ	24
	YP = GP/2.0	TREQ	25
25	IF (IPRINT(2) .EQ. ITERATN) WRITE(6,8100)IV,IMU,NTRY,PHIN,VPP1,	TREQ	26
	2 YP,VPM1,GP,GPM1	TREQ	27
	RETURN A	TREQ	28
	4998 FORMAT(1H * - - - - PROGRAM STOPPED BECAUSE ROLL FAILED TO CONV	TREQ	29
	2ERGE WITHIN 10 ITERATIONS - - - - *)	TREQ	30
30	8100 FORMAT(1X,3I5,6F10.3)	TREQ	31
	8101 FORMAT(1//,3X,*,IV*,3X,*,IMU*,2X,*,NTRY*,4X,*,PHIN*,6X,*,YP*,1*,7X,*,YP*	TREQ	32
	2,7X,*,YP-1*,7X,*,GP*,7X,*,GP-1*/)	TREQ	33
	END	TREQ	34

SUBROUTINE BRWSSP 74/74 OPT=0 ROUND=0/ TRACE FTH 4.6+460 11/26/79 11.25.30

1	SUBROUTINE BRWSSP (N,SIGMH,TMODAL,W,S)	BNSS	2
	C=====	BNSS	3
	C COMPUTES BRETSCHNEIDER WAVE SLOPE SPECTRUM	BNSS	4
	REAL K	BNSS	5
5	DIMENSION 4(N),S(N)	BNSS	6
	DATA A,B,PI,G /467.0626,1940.2444,3.1415927,32.1725/	BNSS	7
	C (TMODAL = 2.76 * SIGMH**.5)	BNSS	8
	TMODAL4 = TMODAL**4	BNSS	9
	C=====	BNSS	10
10	C FOR PIERSON-MOSKOWITZ SPECTRA - TMODAL**4 = 56.8936 * SIGMH**2	BNSS	11
	CON1 = A * SIGMH**2 / TMODAL4	BNSS	12
	CON2 = B / TMODAL4	BNSS	13
	DO 10 I=1,N	BNSS	14
	W4 = W(I)**4	BNSS	15
15	W5 = W(I) * W4	BNSS	16
	ARG = CON2 / W4	BNSS	17
	IF (ARG .GT. 500.) S(I) = 0.	BNSS	18
	IF (ARG .GT. 500.) GO TO 10	BNSS	19
	C=====	BNSS	20
20	C BRETSCHNEIDER WAVE HEIGHT SPECTRUM	BNSS	21
	S(I) = CON1/W5 * EXP(-ARG)	BNSS	22
	C=====	BNSS	23
	C WAVE NUMBER IN DEGREES	BNSS	24
	K = 360.*W(I)*W(I) / (2.*PI*G)	BNSS	25
25	C=====	BNSS	26
	C BRETSCHNEIDER WAVE SLOPE SPECTRUM	BNSS	27
	S(I) = K*K * S(I)	BNSS	28
	10 CONTINUE	BNSS	29
	RETURN	BNSS	30
30	END	BNSS	31

SUBROUTINE ALGRNG 74/74 OPT=0 ROUND=0/ TRACE

FTN 4.6+660

11/26/79 11.25.38

1	C	-----VERSION 3 - CDC 6700 - A L G R N G - JANUARY, 1974-----	ALGR	2
	C	-----S. BALES-----	ALGR	3
	C		ALGR	4
5	C	SUBROUTINE ALGRNG (N,M,S,AREA)	ALGR	5
	*		ALGR	6
	*	*THIS SUBROUTINE COMPUTES THE AREA UNDER THE CURVE FOR A PARTICULAR	ALGR	7
	*	*SPECTRUM. AN ODD NUMBER OF POINTS (FREQUENCIES) SHOULD BE USED.-----	ALGR	8
	*		ALGR	9
10		INTEGER ERROR	ALGR	10
		DIMENSION 4(N),S(N)	ALGR	11
		DATA ITAG/0/	ALGR	12
		DATA EPS/0.0000000001/	ALGR	13
	C		ALGR	14
15		ERROR = 10	ALGR	15
		IF (ERROR.EQ. 10) ITAG = 0	ALGR	16
		ERROR=0	ALGR	17
		M0=M(1)-.03	ALGR	18
		AREA0 = 0.5*S(1)*(M(1)-M0)	ALGR	19
20		MN=M-2	ALGR	20
		AREA=0.	ALGR	21
		TEMP = 0.	ALGR	22
		AREA2 = AREA3 = 0.	ALGR	23
		NOMEGA = 100(M+2)	ALGR	24
25		DO 20 M=1,MN,2	ALGR	25
		A=M(M+2)-M(1)	ALGR	26
		B=M(M+2)-M(M+1)	ALGR	27
		C=M(M+1)-M(1)	ALGR	28
		PAREA = A*A/6.*(S(M)*(3.*C-A)/(A*C)+S(M+1)*A/(B*C)+	ALGR	29
30		2 S(M+2)*(2.*A-3.*C)/(A*B))	ALGR	30
		TEMP = PAREA	ALGR	31
		IF (PAREA.LT. 0.) TEMP = 0.	ALGR	32
		AREA = AREA + TEMP	ALGR	33
		IF (PAREA.GE. 0.) GO TO 20	ALGR	34
35		IF (-PAREA.GT. 0.10*AREA) ERROR=1	ALGR	35
	20	CONTINUE	ALGR	36
		PAREA = 0.	ALGR	37
		IF (AREA..E. 0.000010) GO TO 100	ALGR	38
	C		ALGR	39
40	C	-----SEARCH FOR SPECTRAL CLOSURE-----	ALGR	40
	C		ALGR	41
		TEMP = 0.	ALGR	42
		SMAX = S(1)	ALGR	43
		ITEST = 1	ALGR	44
45		DO 30 I=2,M	ALGR	45
	30	IF (S(I).GT. SMAX) SMAX = S(I)	ALGR	46
		X10PRCN = 0.10*SMAX	ALGR	47
	50	ITEST = ITEST + 1	ALGR	48
		IF (ITEST.EQ. 2) J=1	ALGR	49
		IF (ITEST.EQ. 3) J=M	ALGR	50
50		IF ((SMAX-S(J)).LE. EPS) ERROR = ITEST	ALGR	51
		IF (ERROR.EQ. 2) .OR. (ERROR.EQ. 3)) ITAG = 1	ALGR	52
		IF (ERROR.EQ. 2) .AND. (ITEST.EQ. 2)) TEMP=AREA+AREA0	ALGR	53
		IF (ERROR.EQ. 2) .AND. (ITEST.EQ. 2)) AREA=TEMP	ALGR	54
55		IF (S(J).GT. X10PRCN) ERROR = ITEST + 2	ALGR	55
		IF (ERROR.GT. 0) .AND. (ITAG.EQ. 0)) ITAG = 1	ALGR	56
		IF ((J.EQ. N) .AND. (ERROR.EQ. 4)) GO TO 107	ALGR	57
			ALGR	58

SUBROUTINE ALGRNG

74/74

OPT=0 ROUND=0/ TRACE

FTN 4.6+468

11/26/79 11.25.38

	IF (IJ.EQ. N) .AND. (ERROR.LT. 5)) GO TO 108	ALGR	59
	IF (ERROR.EQ.4) .OR. (ERROR.EQ. 5))GO TO 68	ALGR	60
68	IF (ITEST.LT. 3) GO TO 50	ALGR	61
	GO TO 108	ALGR	62
	C	ALGR	63
	C-----DRAW A STRAIGHT LINE THRU FIRST (LAST) TWO SPECTRAL VALUES	ALGR	64
	C-----TO THE ABSXISSA AND ADD ON AREA FOR CLOSURE AT LOW (HIGH)	ALGR	65
65	C-----FREQUENCY END.-----	ALGR	66
	C	ALGR	67
	60 IF (ERROR.EQ.5) .AND. (S(I).GE. S(J-1)))GO TO 75	ALGR	68
	IF (ERROR.EQ.4) .AND. (S(I).GE. S(2)))GO TO 50	ALGR	69
	IF (ERROR.EQ. 4) J=2	ALGR	70
70	IF (ERROR.EQ. 5) J=N	ALGR	71
	SLOPE = (S(J-1) - S(I))/(N(IJ-1) - N(IJ))	ALGR	72
	IF(SLOPE .LE.0.) GO TO 70	ALGR	73
	SLOPE = AMIN1(-SLOPE,-1.0)	ALGR	74
	IF (IJ.EQ. 2) J = 1	ALGR	75
75	ANEN = -0.5 * S(I)**2 / SLOPE	ALGR	76
	IF (ERROR.EQ. 4) AREA2 = AMIN1(AREA0,ANEN)	ALGR	77
	IF (ERROR.EQ. 5) AREA3 = ANEN	ALGR	78
	75 TEMP = AREA + AREA2 + AREA3	ALGR	79
	IF (IJ.LT. N) GO TO 50	ALGR	80
80	107 TEMP = AREA + AREA2 + AREA3	ALGR	81
	AREA = TEMP	ALGR	82
	108 RETURN	ALGR	83
	C	ALGR	84
	END	ALGR	85

SUBROUTINE FINSTAB 74/74 OPT=0 ROUND=0/ TRACE

FTN 4.6+468

11/26/79 11.25.38

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1      SUBROUTINE FINSTAB (IS), RETURNS (AAA,988)
      *
      *SUBROUTINE TO PREDICT STABILIZED (FIN) ROLL MOTION USING CONOLLY AND
      *COX.
5      *
      REAL MM,KK1,KK2,KK3,KR,KI,K1,K2,K3
      COMMON/ITRTN/STAT,PRCN,NTRY,IMU,KNU,IV,GP,GPM1,YPM1,YP,YPP1,
2      PHIN,IPRINT(8),ITERATE,MPMI
      COMMON /STAB/ NSTAB,M(10),AREA(10),R(10),OCLDBFS(10),M0(10),
10     2 M1(10),M2(10),M3(10),M4(10),GK(10),GV(10),K1(10),K2(10),K3(10),
      2 A1(10),A2(10),A3(10),B1(10),B2(10),B3(10),DAMPU(13),DAMPS(10,13),
      2 DUC(5,6),SSIGLC(10,13),MMU,ILC,ISC,WE(40,35),
      2 SSIGVLC(10,13),SSIGVSC(10,13),
15     2 SUR(40,35),M(40),MM,COSI(35),CON1,SSIGSC(10,13),
      2 VFS(5),DISPLB,GM , BMOTLC(10,13),
      2 BMOTSC(10,13),BVELSC(10,13),BVELLC(10,13),BSTOP(5),BVELMAX(5)
      2 ,VSAT,ITEST(10,13)
      DIMENSION SMOUR(35),SSU(40,35),FINM(40,35),FINV(40,35),BMOUR(35),
20     2 BVMOUR(35),SSV(40,35),SVMOUR(35)
      DATA RMO/1.99/ , PI/3.1415926/
      *
      *INITIALIZE.
      *
      N=IS
      MM=M(N)
25     AAREA=AREA(N)
      RR=R(N)
      OCLDBF=OCLDBFS(N)
      MM0=M0(N)
30     MM1=M1(N)
      MM2=M2(N)
      MM3=M3(N)
      MM4=M4(N)
      GK=GK(N)
35     GV=GV(N)
      KK1=K1(N)
      KK2=K2(N)
      KK3=K3(N)
      AA1=A1(N)
40     AA2=A2(N)
      AA3=A3(N)
      BB1=B1(N)
      BB2=B2(N)
      BB3=B3(N)
45     CON3=RMO*VFS(IV)*VFS(IV)*MM*AAREA*RR/(DISPLB*GM)
      SSN = 1.0
      ISOK = 0
75     DAMPS(N,IMU) = DUC(IV,1)
      IF (ITERATE .EQ. 0) GO TO 90
50     NTRY = 0
      YP = 0.0
      NTRY = NTRY + 1
      X = YP
80     DAMPS(N,IMU) = DUC(IV,1)+1.61*DUC(IV,2)*X**0.772 + 1.88*DUC(IV,3)
55     2)*X + 4.00*DUC(IV,4)*X**2.0 + 9.48*DUC(IV,5)*X**3.0 +
      2 24.0*DUC(IV,6)*X**4.0
90     PHIS = SSIGVLC(N,IMU) = SSIGVSC(N,IMU) = 0.

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SUBROUTINE FINSTAB 74/74 OPT=0 ROUND=0/ TRACE

FTN 4.6+460

11/26/79 11.25.30

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      BMOTSC(IN,INU)=BVELSC(IN,INU)=BMOTLC(IN,INU)=BVELLC(IN,INU) = 0.      FST      59
      *                               FST      60
60    *BEGIN LOOP OVER SPREADING ANGLE (INU).      FST      61
      *                               FST      62
      IF (ISC.EQ. 0) NNU = 1      FST      63
      DO 400 INU = 1,NNU      FST      64
      *                               FST      65
65    *BEGIN LOOP OVER WAVE FREQUENCY.      FST      66
      *                               FST      67
      DO 200 IN=1,NW      FST      68
      WNE=WE(IN,INU)      FST      69
      WESQ=WNE*WNE      FST      70
70    TUNF = WE(IN,INU)/WPHI      FST      71
      BA = 2.*DAMPU(INU)*TUNF      FST      72
      A = 1. - TUNF*TUNF      FST      73
      CA = SQRT(1+A + BA*BA)      FST      74
      *                               FST      75
75    *COMPUTE EFFECTIVE LIFT CURVE SLOPE.      FST      76
      *                               FST      77
      DCLDBE = H40 + H41*WNE + H42*WESQ + H43*WESQ*WNE + H44*WESQ*WESQ      FST      78
      DCLDBE = DCLDBE*DDCLDBF*180./3.1415926      FST      79
      *                               FST      80
80    *COMPUTE AMPLITUDE OF FIN ANGLE TO STABILIZED ROLL.      FST      81
      *                               FST      82
      KR = KK1 - WESQ*KK3      FST      83
      KI = WNE*KK2      FST      84
      BR = BB1 - WESQ*BB3      FST      85
      BI = WNE*BB2      FST      86
85    CON4=KR*KR + KI*KI      FST      87
      CON5 = BR*BR + BI*BI      FST      88
      BAONS = GGV*GGV*SQRT(CON4/CON5)      FST      89
      *                               FST      90
90    AR = AA1 - WESQ*AA3      FST      91
      AI = WNE*AA2      FST      92
      *                               FST      93
      BS = 2.*DAPSI(IN,INU)*TUNF      FST      94
      CS = SQRT(1+A + BS*BS)      FST      95
95    CON6 = AR*BR - AI*BI      FST      96
      CON7 = AR*BI + AI*BR      FST      97
      CON8 = AR*AR + AI*AI      FST      98
      SAONCSP = SSN*CON3*DCLDBE*BAONS/(CS*SQRT(CON8))      FST      99
      *                               FST      100
100   COSKSI = (KR*CON6 + KI*CON7)/SQRT(CON4*CON8*CON5)      FST      101
      SINKSI = (KI*CON6 - KR*CON7)/SQRT(CON4*CON8*CON5)      FST      102
      *                               FST      103
      *COMPUTE ROLL REDUCTION FACTOR FOR THIS FREQUENCY.      FST      104
      *                               FST      105
105   SONU = CA/CS/SQRT(1.+2.*SAONCSP*((1+COSKSI+      FST      106
      2 BS*SINKSI)/CS)+SAONCSP*SAONCSP)      FST      107
      *                               FST      108
      *COMPUTE STABILIZED ROLL , FIN MOTION, AND FIN VELOCITY SPECTRA.      FST      109
      *                               FST      110
110   SSU(IN,INU) = SUR(IN,INU) * SONU * SONU      FST      111
      SSV(IN,INU) = SSU(IN,INU)*WESQ      FST      112
      FINV(IN,INU) = SSU(IN,INU) * BAONS * BAONS / CON8      FST      113
      FINN(IN,INU) = FINV(IN,INU) * WESQ      FST      114
      *                               FST      115

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115	SEND OF LOOP OVER WAVE FREQUENCY.	FST	116
*		FST	117
	200 CONTINUE	FST	118
*		FST	119
	*DETERMINE RMS ROLL, FIN MOTION, AND FIN VELOCITY VALUES.	FST	120
120	*	FST	121
	CALL ALGRN3 (NW,N,SSU(1,INU),SMOUR(INU))	FST	122
	CALL ALGRN3 (NW,N,SSV(1,INU),SVNOUR(INU))	FST	123
	CALL ALGRN3 (NW,N,FINM(1,INU),BMOUR(INU))	FST	124
	CALL ALGRN3 (NW,N,FINV(1,INU),BVNOUR(INU))	FST	125
125	*	FST	126
	*STORE LONGCRESTED VALUES (NO ITERATION OVER ROLL DAMPING IS DONE).	FST	127
*		FST	128
	IF (ISC.NE. 0) GO TO 210	FST	129
	SSIGLC(N,INU) = SORT(SMOUR(INU))	FST	130
130	SSIGVLC(N,INU) = SORT(SVNOUR(INU))	FST	131
	BMOTLC(N,INU) = SORT(BMOUR(INU))	FST	132
	BVELLC(N,INU) = SORT(BVNOUR(INU))	FST	133
	IF (ITERATE.EQ. 0) GO TO 400	FST	134
	SSGVWLC = SSIGVLC(N,INU)/WPHI	FST	135
135	PHIN = DAMPS(N,INU)	FST	136
	CALL ITREQ(SSGVWLC), RETURNS(00,400,900)	FST	137
*		FST	138
	*BEGIN SUMMING OF SHORTCRESTED RESPONSE DATA.	FST	139
*		FST	140
140	210 PHIS = PHIS + COS1(INU)*SMOUR(INU)	FST	141
	SSIGVSC(N,INU) = SSIGVSC(N,INU) + COS1(INU)*SVNOUR(INU)	FST	142
	BMOTSC(N,INU) = BMOTSC(N,INU) + COS1(INU)*BMOUR(INU)	FST	143
	BVELSC(N,INU) = BVELSC(N,INU) + COS1(INU)*BVNOUR(INU)	FST	144
	*	FST	145
145	SEND OF LOOP OVER MU.	FST	146
*		FST	147
	400 CONTINUE	FST	148
	IF (ISC.EQ. 0) GO TO 800	FST	149
	PHIS = SORT(CON1*PHIS)	FST	150
150	SSIGVSC(N,INU) = SORT(CON1*SSIGVSC(N,INU))	FST	151
	BMOTSC(N,INU) = SORT(CON1*BMOTSC(N,INU))	FST	152
	BVELSC(N,INU) = SORT(CON1*BVELSC(N,INU))	FST	153
	IF (ITERATE.EQ. 0) GO TO 250	FST	154
	SSGVW = SSIGVSC(N,INU)/WPHI	FST	155
155	PHIN = DAMPS(N,INU)	FST	156
	CALL ITREQ(SSGVW), RETURNS(00,250,900)	FST	157
250	SSIGSC(N,INU) = PHIS	FST	158
	IF ((INSAT.EQ.0).OR. (ISOK.EQ.1)) GO TO 800	FST	159
	X1 = (BSTOP(IV)/BVELMAX(IV))*(BVELSC(N,INU)/BMOTSC(N,INU))	FST	160
160	X2 = SIN(X1)	FST	161
	F = (4/PI)*(X2/X1)	FST	162
	Y = BSTOP(IV)/(BMOTSC(N,INU)*1.41421)	FST	163
	SSH = 1/(1 - X2)*(1 - F*X2)*ERRF(Y) + (F - 1)*X2*ERRF(Y/X2)	FST	164
	IF (SSH.GE. .98) GO TO 870	FST	165
165	ISOK = 1	FST	166
	GO TO 75	FST	167
	870 ITEST(N,INU) = 1H*	FST	168
	800 RETURN AAA	FST	169
	900 RETURN BBB	FST	170
170	END	FST	171

FUNCTION ERRF		74/74	OPT=0 ROUND=0/ TRACE	FTN 4.6+60	11/26/79	11.25.30
1	C	REAL FUNCTION ERRF(X)			ERF	2
		DATA P/.47047/ ,C1/.34802/ ,C2/-.09500/ ,C3/.74786/			ERF	3
		IF (ABS(X) .GT. 15.) GO TO 5			ERF	4
5		T = 1/(1 + P*X)			ERF	5
	C				ERF	6
		ERRF = 1 - (C1*T + C2*T*T + C3*T*T*T)*EXP(-X*X)			ERF	7
	C				ERF	8
		GO TO 10			ERF	9
10	5	ERRF = 1.			ERF	10
	10	RETURN			ERF	11
		END			ERF	12
					ERF	13

APPENDIX C

SPECIAL ALGORITHMS

ITERATION OVER ROLL-ROLL DAMPING EQUATIONS*

If roll damping coefficient n is independent of roll angle (as, for example, in some cases where bilge keels are not appended to the hull), FINCON executes with a unique value of n for each required ship speed. However, as has often been found through model experiments, roll damping is dependent on roll angle, as well as ship speed and natural roll frequency. Thus, the program requires a different input description of roll damping. As described in the section of this report entitled Program Input, n is, in this case, defined by

$$n_{IR} = d_0 + 1.61 d_q y^{0.772} + 1.88 d_1 y + 4 d_2 y^2 + 9.4 d_3 y^3 + 24 d_4 y^4 \quad (7)$$

where $y = \sigma_{\dot{\phi}}/\omega_{\phi}$ and the fractional power $q = 0.772$ arises from the turbulent skin friction contribution.

Thus, the computational problem is then to solve the roll equation of motion involving $\dot{\phi}$ for the correct value of n_{IR} .** This has been accomplished by finding the intersection of the known curve of Equation (7) (i.e., $n_{IR} = f(y)$ where $y = \sigma_{\dot{\phi}}/\omega_{\phi}$) and the initially unknown curve $g = g(n_{IR})$, which is computed from the solution of the roll-rate equation of motion, using the roll damping value n_{IR} .

The solution is found, in brief, by the following procedure as applied to the example of Figure 4:

1. Assume $y_0 = 0$, determine $n_0 = f(0)$
2. Using n_0 , solve the roll-rate equation of motion for

$$g_0 = \frac{\sigma_{\dot{\phi}}}{\omega_{\phi}} = g(n_0)$$

*Taken from future report already under preparation by Cox.

**The subscript IR is used here since it is assumed that motion is taking place in irregular waves.

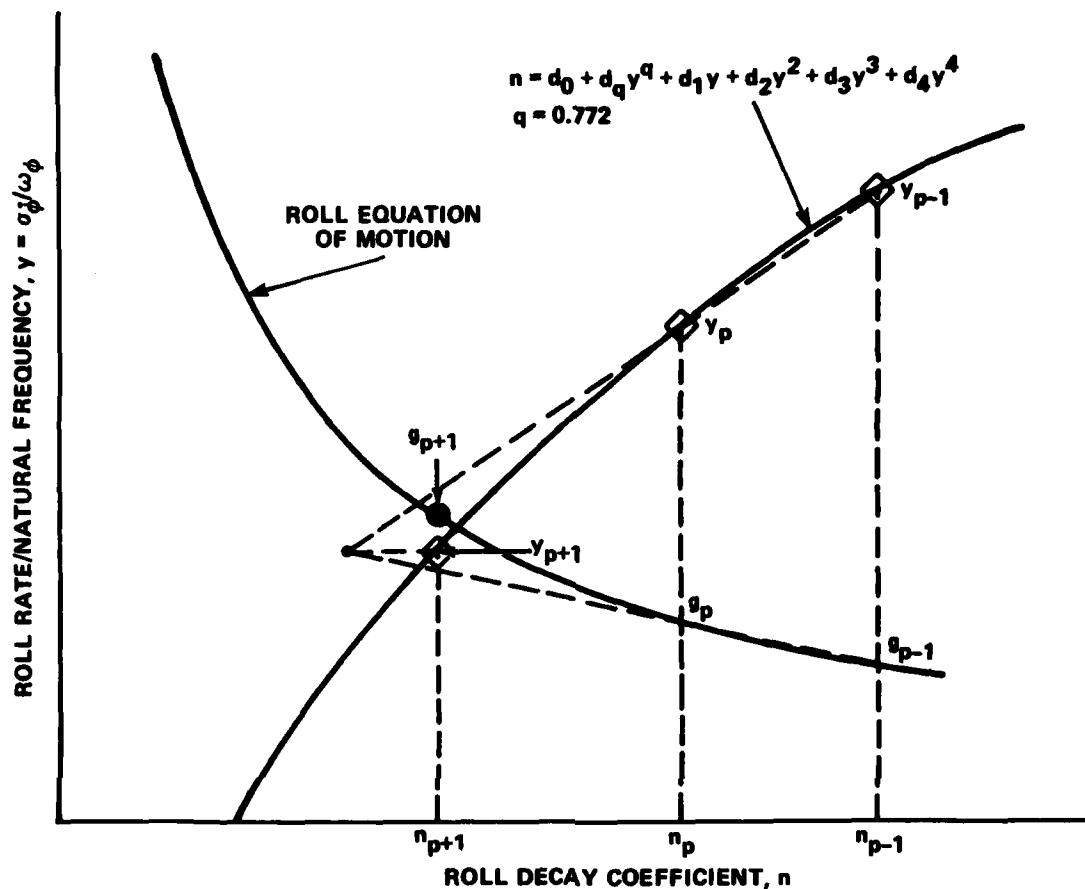


Figure 4 - Illustration of Iteration over Nonlinear Roll-Roll Damping

3. Assume $y_1 = g_0/2$, and determine $n_1 = f(y_1)$
4. Compute $g_1 = g(n_1)$
5. For $p = 2, 3, 4 \dots$ etc.
 Calculate $y_p = (g_{p-2}y_{p-1} - g_{p-1}y_{p-2}) / [(y_{p-1} - y_{p-2}) - (g_{p-1} - g_{p-2})]$
 and determine $n_p = f(y_p)$
6. Compute $g_p = g(n_p)$ and test if $|1 - g_p/y_p| \leq \epsilon$ where ϵ is some small number. If the test is not satisfied, proceed to the next

step of the iteration. If the test is satisfied, then the iteration is terminated and n_p , g_p is the required solution.

In FINCON, ϵ is usually taken as 0.01 and is called PRCN in the program itself. This means convergence is attained when $|1 - g_p/y_p| \leq 0.01$. Table 6 provides a typical printout of the steps when IPRINT(2) is input as ITERATN. IV indicates the speed, such as the first; IMU indicates the heading, such as the seventh or 90 degrees; NTRY indicates the numbers of attempts; PHIN indicates the latest selected n value; YP+1 the latest calculated roll-rate value (divided by ω_ϕ); and g_{p-1} , y_{p-1} , g_{p-2} , and y_{p-2} are the previous calculated roll-rate values (divided by ω_ϕ), going back in time. It should be noted that the printing occurs at various steps within each repetition of the iteration rather than just at the end (or beginning) of each attempt. In the particular case listed in Table 6, the first seven lines refer to the unstabilized case, while the last five refer to the stabilized case. Also, it is recalled that the iteration is operating over either the long-crested or short-crested RMS roll rate. A similar procedure, implemented in the Navy Standard Ship Motion Program (SMP-79) (six-degrees-of-freedom) operates over the resonant region of the (long-crested) roll transfer function. Also, the roll-rate values actually used here for the internal testing (e.g., in step 5 above) are usually taken to be the statistic which corresponds to the experimental data described by Equation (7). In all cases, the RMS, single-amplitude roll rate is used. The second page of Table 4 provides a further illustration of this example.

GENERALIZATION OF COSINE SQUARED LAW FOR SHORT-CRESTED SEAS

In general, a 15-degree cosine square spreading function about ± 90 degrees is used for calculation of ship motions in short-crested or multi-directional seas (e.g., see Reference 3). However, it has been found in calculations done previously, that a more refined angular spread may be required for roll-motion calculations at higher ship speeds. This is due to the highly tuned nature of roll motion. That is, when considering RMS

roll motion across all ship headings at high speeds, the maximum values may occur in between the standard 15-degree increments of heading; and, if not considered, some loss of energy may be noticed.

Therefore, a FINCON program option is available for varying the spreading angle used. Spreading angles of 5, 10, or 15 degrees may be specified on data card 9. Though the spreading is thus varied, the predominant heading angles are never varied from the usual 15-degree increments. Table 4 presents results when a 15-degree spreading is specified. In general, either a 15-degree spreading (e.g., for lower speeds) or a 5-degree spreading (e.g., for higher speeds) is recommended; although no specific guidelines for the use of either are currently available. A major difference between the two angles is, of course, in program run time and, thus, in cost. A typical 15-degree spreading run may cost as much as 68 percent less than a 5-degree run. In general, 10-degree spreadings are not recommended, because irregular trends across ship headings may be perceived. For example, for a 75-degree predominant direction, the 10-degree spreading angles are at -15, -5, 5, . . . , 155, 165 degrees; and, for a 90-degree predominant heading, they are at 0, 10, 20, . . . , 165, 180 degrees; while for 105 degrees they are at 15, 25, 35, . . . , 185, 195 degrees. If roll is highly tuned, then adjacent short-crested RMS roll angles may appear to have erratic behavior (e.g., decrease at 75 degrees, increase at 90 degrees, decrease at 105 degrees, etc.). This is due, numerically, just to the difference in the base spreading angles in each case.

The algorithm implemented in FINCON to provide this generalization in the short-crested procedure is defined by the following:

1. Let I be the specified spreading angle (5, 10, or 15 degrees) or angle of constant energy
2. Set $\ell = (180/I)/2$
and $c = 1/\ell$
3. Then,

$$\sigma_{sc}^2(\mu) = c \sum_{p=-(\ell-1)}^{(\ell-1)} \sigma_{\ell cn}^2\left(\mu + \frac{p\pi}{I}\right) \cos^2 \frac{p\pi}{I}$$

where σ_{lc}^2 = squared long-crested RMS roll or variance
 μ = predominant heading angle

σ_{sc}^2 = squared short-crested RMS roll or variance.

The only interval required by the algorithm is I, the angular interval over which a constant wave energy-ship response is assumed.

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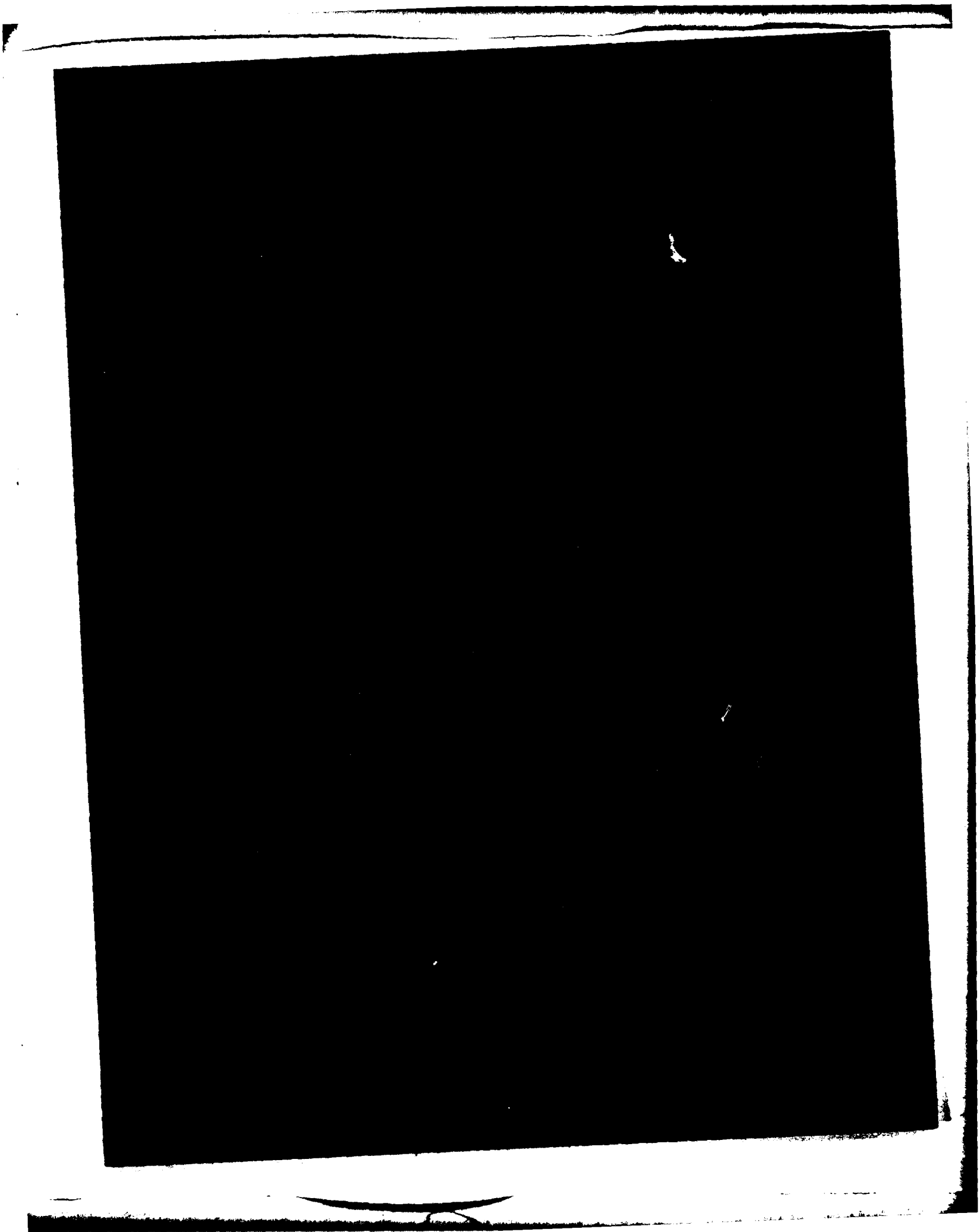
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